HUMAN RESOURCES RESEARCH ORGANIZATION ALEXANDRIA VA
TRAINING PROGRAMS FOR THE SYNTHETIC FLIGHT TRAINING SYSTEM. (U)
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DAHC19-73-C-00 AD-A033 587 F/6 5/9 DAHC19-73-C-0004 HUMRRO-FR-CD(D)-75-3 UNCLASSIFIED NL 1 oF 2 AD A033587

FR-CD(D)-75-3

Final Report



Training Programs for the Synthetic Flight Training System

Oran B. Jolley, Robert N. Isley, and Edward J. Miller

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Approved for public release; distribution unlimited.

June 1975

Prepared for

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The findings, guidelines, and assumptions of this report are not to be construed as an official Department of the Army acceptance or position. The Army Research Institute for the Behavioral and Social Sciences caution the reader that the techniques described herein have not, as yet, undergone adequate evaluation. The contents of the report reflect the views of the Human Resources Research Organization which is responsible for the accuracy and completeness of the material thus far presented. The report does not necessarily reflect the opinion of the U.S. Army Avistion Center.

CURITY CLASSIFICATION OF THIS When Detagentered	READ INSTRUCTIONS
REPORT DOCUMENTATION PAGE	BEFORE COMPLETING FORM
REPORT NUMBER 2./GOVT ACCESSION	NO. 3. RECIPIENT'S CATALOG NUMBER
FR-CD(6)-75-3	
TITLE (and Subtitle)	TYPE OF REPORT & PERIOD COVERED
Training Programs for the Synthetic Flight	Final Man 2+
Training System.	Final Kepert,
(4) HumRR	FR-CD(D)-75-3
AUTHOR(a)	6. CONTRACT OR GRANT NUMBER(*)
Oran B./Jolley, Robert N./Isley Edward J.	(15)
Miller	
PERFORMING ORGANIZATION NAME AND ADDRESS	DAHC 19-73-C-20064
Human Resources Research Organization (HumRRO)	REA & WORK UNIT NUMBERS
300 North Washington Street Alexandria, Virginia 22314	16
Alexandria, Virginia 22314 🗸 400	T20764715A757
CONTROLLING OFFICE NAME AND ADDRESS	IL. REPORT DAY
U.S. Army Research Institute for the Behavioral	4 001 075
and Social Sciences, 1300 Wilson Boulevard,	13. NUMBER OF PAGES 121 pages
Arlington, Virginia 22209 MONITORING AGENCY NAME & ADDRESS(II different from Controlling Office)	
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(2)133Po	Unclassified
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a discussion of the present status of the automated training program feature of Device 2B24.

The training program described in the report has not been subjected to a full-scale evaluation. Portions of the program were tested, using HumRRO and U.S. Army volunteers, and was found suitable for further use. The draft training materials developed for this program are documented in this report in the event an evaluation of the program is undertaken in the future.

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FINAL REPORT

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by

Oran B. Jolley, Robert N. Isley, and Edward J. Miller

June 1975

Work Unit SYNTRAIN

Prepared for

U.S. Army Research Institute for the Behavioral and Social Sciences Contract DAHC19-73-C-0004

HUMAN RESOURCES RESEARCH ORGANIZATION 300 North Washington Street • Alexandria, Virginia 22314

FOREWORD

This Final Report by the Human Resources Research Organization summarizes final projects of Work Unit SYNTRAIN performed for the U. S. Army Research Institute for the Behavioral and Social Sciences. The purpose of Work Unit SYNTRAIN has been to expedite the application of advances in training technology to the design and utilization of Army aviation synthetic training equipment.

Conducted at Fort Rucker, Alabama, by the Humrro Central Division,
Work Unit SYNTRAIN began in 1968 and is terminated with the publication
of this report. Principal SYNTRAIN reports are Technical Report 70-10,
Device-Task Fidelity and Transfer of Training: Aircraft Cockpit Procedures Training; Technical Report 72-11, Determining Training Device
Requirements in Fixed Wing Aviator Training; Technical Report 73-20,
Research on Synthetic Training: Device Evaluation and Training Program
Development; and Technical Report 75-12, Mission Suitability Testing
of an Aircraft Simulator.

Mr. Robert N. Isley served as Principal Investigator during the last year of the SYNTRAIN effort. In addition to the authors, a number of Humrro personnel, both past and present, have made significant contributions to the research reported here. The efforts of Dr. Paul W. Caro, Simulation Programs Manager for Humrro and the initial Principal Investigator for SYNTRAIN, Dr. Melvin D. Montemerlo, Mr. Eugene R. Hall, Dr. John L. Bilbrey, Jr., Mr. Maurice Siskel, Jr., and Mr. Winon E. Corley were particularly helpful.

Military support for Work Unit SYNTRAIN was provided by the U. S. Army Research Institute Human Research Unit, Fort Rucker, Alabama.

LTC Donald E. Youngpeter was Unit Chief during the period covered by this report.

HumRRO research for the Department of the Army is conducted under Contract No. DAHC 19-73-C-0004. Army training research is performed under Army Project No. 20764715A757. The Contracting Officer's Technical Representative for Contract No. DAHC 19-73-C-0004 is Dr. Arthur J. Drucker.

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Chapter 1

INTRODUCTION

BACKGROUND

Work Unit SYNTRAIN, Modernization of Synthetic Training in Army

Aviation, was initiated in 1968 to expedite the application of advances
in training technology to the design and utilization of Army aviation

synthetic training equipment through surveys of training device design

requirements and technology and by human factors and training research.

Major aspects of the Work Unit have already been reported in the following technical reports: Device Task Fidelity and Transfer of Training:

Aircraft Cockpit Procedures Training; Determining Training Device

Requirements in Fixed Wing Aviator Training; Research on Synthetic Training: Device Evaluation and Training Program Development; and Mission

Suitability Testing of an Aircraft Simulator. 1

The last report cited above dealt with the results of an operational suitability test of the Army's simulator for the UH-1H aircraft, Device 2B24. Device 2B24 is the first subsystem of a series of flight simulators, collectively called the Synthetic Flight Training System (SFTS), envisioned for the Army (subsystems for the CH-47C, Device 2B31, and the AH-1Q, Device 2B33, are currently in procurement²). The operational

A complete listing of Work Unit SYNTRAIN publications and research by-products is given in Appendix A.

HumRRO has been involved in the design and development of both these subsystems since their conception, and both will have several advanced design-for-training features intended to enhance their training value. Advanced features that will require testing and evaluation include: (a) automated maneuver demonstration systems; (b) sling load training system; (c) weapons firing and scoring system; (d) visual transition flight training capability, and (e) pilot and gunner flight instructor stations.

suitability test of Device 2B24 was conducted to evaluate the device's potential for cost-effectively accomplishing the instrument phase of Army rotary wing flight training and facilitating UH-1 transition training, aviator proficiency evaluation, standardization, and flight training quality control. Test results confirmed the device's mission suitability regarding both cost-effectiveness and transfer of training (1, 2, 3). Forty-three hours of simulator time and approximately 6 hours of aircraft time were substituted for 60 hours of aircraft and 26 hours of training device time, with no degradation of student pilot performance. The Army obtained similar results in a subsequent test of the training value of Device 2B24 (4). The 2B24 is currently being used routinely to support the instrument flight training phase of the undergraduate flight training program, the Rotary Wing Qualification Course, the Rotary Wing Instrument Flight Examiner Course, and proficiency training for faculty and staff at the Aviation Center.

Having established Device 2B24's capability of fulfilling its training mission at considerable cost savings, attention turned to two areas of research: (a) examination of the automated training program features of the device which were not fully assessed during the suitability test, and (b) development of a training program to enable more extensive utilization of the 2B24 with undergraduate trainees. Earlier test findings had indicated that the device's design was such that even greater potential utilization might be achieved in the undergraduate flight training program (3).

DEVICE 2B24 AUTOMATED TRAINING PROGRAMS

The developmental nature of the 2B24 automated training programs supplied by the manufacturer and the problems associated with their use have been discussed in some detail in the report of the operational suitability test cited earlier (1, 2). Essentially, the automated training exercises developed for Device 2B24 are analogous to computer-administered instruction (CAI) programs, in that a number of revisions are typically required before they can be considered appropriate for routine training use. The suitability test results indicated that, while considerable additional developmental effort would be required in this area, such effort not only would likely enhance the device's training effectiveness, but also would add to the technology of automated training. Consequently, these developmental efforts were continued, on a time-available basis, in cooperation with the Army's device programing staff. These efforts and the present status of the automated training programs are discussed in Chapter 2.

FURTHER UTILIZATION OF DEVICE 2B24

Beginning in 1973, research activities were undertaken by HumRRO to design and evaluate a training program that would maximize Device 2B24's potential contribution to undergraduate flight training. The resulting experimental training program, designed to replace the undergraduate flight training program currently in use, is conceptually and administratively different from the traditional approach to undergraduate flight training. The procedures involved in its design were similar to those employed in earlier HumRRO research (5, 6), in which certain methods

and techniques were added, dropped, or extensitvely modified during the program's evolution.

Unlike those previous developmental efforts, however, a step vital in the development of any successful training program was not performed: program testing. This step is essential to enable program refinement and evaluation, but was not possible to accomplish during the time available due to other activities at the Aviation Center. Therefore, even though portions of the experimental training program were tried out by Humra personnel and Army volunteers, it is by no means complete, and is described here only as the basis for future development.

In addition, a number of administrative and/or policy decisions concerning the program have not been addressed. Major among these are the extent of the requirements for solo flight in the UH-1 and the possible need for revisions to both academic and in-flight examinations to make course administration more efficient. Had the proposed test of the new program been carried out, guidance for making these decisions would have been developed.

Discussion of the experimental training program appears in Chapter 3.

Chapter 2

AUTOMATED TRAINING PROGRAMS

Device 2B24 represents the first major acquisition of state of-the-art helicopter flight simulators for the U. S. Army. Incorporated into the device are the latest concepts in training device design and training technology. The main concept that distinguishes the 2B24 from existing systems is that of automated training. The manufacturer delivered the device with developmental programs capable of demonstrating the automated training functions of the device. These programs were not intended to be a complete, ready-to-use training program. Ten sample programs were available at delivery, including Basic Instruments, Instrument Takeoff, ADF Approach, VOR Approach, ILS Approach, Holding, Back Course ILS Approach, a Cross-Country Flight, and In-Flight Engine Restart, and an Instrument Checkride.

The ten programs included such features as (a) demonstrations of maneuvers and procedures; (b) programed instructional periods containing automatic briefings, computer-controlled demonstrations, guided practice segments with performance monitoring, and adaptive training segments in which the difficulty level of the task being performed by the student is automatically adapted to the skill level or performance level of the student; and (c) an automated checkride with performance monitoring and a hard-copy printout of all out-of-tolerance conditions. In addition, the device provides a capability for the processing of training data for quality control purposes.

During the operational suitability test, conducted as an integral part of the Expanded Service Test, the automated training features were

examined, and a great deal of useful information was obtained. Unfortunately, the programs were unusable in their existing condition. Due to the relatively long lead time necessary to correct the problems associated with the automatic features and the press of other suitability test activities, further testing and analysis of these automatic features were postponed. After completion of the test and relocation of the device to a permanent facility, the Aviation Center demand for operational training time in the device precluded any further systematic automatic program testing and evaluation. The Center did, as time permitted, analyze, edit, and re-record each automatic briefing. Software modifications were also made to standardize maneuver and procedure demonstrations.

With the receipt of additional devices at Fort Rucker, more time was available for examination of programs and correction of obvious errors and software problems. HumRRO activities were coordinated with responsible personnel from the Flight Simulator Division of the Aviation Center, and a continuous exchange of findings and results ensued. Many problems were encountered which initially made it impossible to complete all the programs. For example, one problem was identified as the need for an adjustment to the tone level on the tape recorder which would cause the tapes and computer to get out of synchronization, totally negating use of the automated programs. Just recently, a procedure was discovered which virtually eliminates this problem. In their present condition, the automated training programs are usable, the audio content is technically correct, and the maneuver demonstrations are procedurally correct. Performance recording printouts, however, are unusable.

HumRRO personnel examined in detail each of the ten automated programs. Based on the findings of this examination, portions of the programs were tried out on fixed wing qualified, non-instrument rated volunteers, and also on several individuals with no flight experience, during the presentation of helicopter instrument instruction to them. The programs were used to provide demonstrations of specific maneuvers at the appropriate time in the training program, and were also used to provide guided individual practice and periods of adaptive training. For the most part, an instructor was present to determine when to introduce the automated program and provide supplemental instructions to clarify program objectives. Occasionally, students were provided training time in the automated mode without an instructor in the cockpit and then were subsequently required to demonstrate their skills and knowledges acquired in this fashion to an instructor. Thus, at the present time the automated programs could be a useful tool for qualified instructors trained to use them. They are not, however, considered ready for routine training use without additional refinement and testing, without which their use by students and instructors would likely be haphazard. Likewise, the absence of refinement and evaluation precluded the planned incorporation of one or more of the automated training programs into the experimental training program described in the next chapter of this report.

Some of the expected benefits from the use of the automated programs are: (a) they perform many of the routine (repetitive) operations here-tofore assigned to the human instructor, freeing him for other instructional tasks; (b) they provide a means of presenting standardized

maneuvers and procedures to all students; (c) they have the capability to provide objective measures of deviations on selected parameters of pilot and aircraft performance from a specified standard; and (d) they have the capability to provide quality control through standardization of Army instrument training programs and to identify areas of non-standard performance by individuals and units.

Because of the contribution these programs could make to standardization of training, both at the Aviation Center and at field locations, a research program to continue the previous developmental efforts would appear warranted. Until such time as the automated training programs are formally tested, their training effectiveness remains unknown.

Chapter 3

THE EXPERIMENTAL TRAINING PROGRAM

RATIONALE

The purpose of the Primary Phase of undergraduate flight training is to enable the student to learn the psychomotor skills necessary to control the attitude of the aircraft in flight, takeoff, landing, and visual navigation. Traditionally, this has been accomplished using the real world as an attitude reference (contact flying). The sequence generally followed in flight training has been (a) day contact flight, (b) night contact flight, and (c) instrument flight. Control manipulation to achieve a desired aircraft attitude, however, is the same, regardless of the cue used, contact or instrument.

As early as the 1930s, the efficacy of the traditional sequence was challenged in an experiment where instruments were taught first (7). This and later studies produced evidence that there is positive transfer of training when instrument flight training is conducted first, followed by contact training, and negative transfer of training when the usual sequence is followed (8, 9, 10). Another form of instruction, known as integrated training—training in which the student is required to learn to perform a maneuver using instrument cues prior to learning it using contact cues during a common flight period—is used by the U. S. Air Force in undergraduate flight training and has been alluded to by the FAA in its private pilot requirements (11, 12).

Based upon prior research and the practices of other training groups, it was hypothesized that Device 2B24 could be used to teach the

beginning student initial flight control skills before he goes to the aircraft, and that such training would facilitate the subsequent learning of contact flight skills. This expectation is testable, and the program of instruction described below was developed for use in a test of this hypothesis.

OVERVIEW

To provide a frame of reference for the discussion of the experimental training program, a brief description of the current program is provided.

Current Undergraduate Flight Training Program

The Officer/Warrant Officer Rotary Wing Aviator Courses (O/WORWAC) at the U. S. Army Aviation Center graduate approximately 90 aviators each month. The split between officers and warrant officer candidates (WOCs) is approximately 50-50, and the courses are identical except that the WOC receives two additional weeks of preflight training, designed to help prepare him to assume the responsibilities of a U. S. Army warrant officer.

The courses are conducted in two phases. Phase I lasts 12 weeks, during which the trainees receive 85 hours of dual flight instruction and solo practice in the TH-55 helicopter. All of this training is conducted in accordance with Visual Flight Rules (VFR), with aircraft control being maintained by reference to real world cues. In addition, the trainee receives 98 hours of academic instruction. Phase I training, both flight and academic, is conducted by a civilian contractor.

Upon successful completion of Phase I, the trainee begins Phase II training, which lasts 20 weeks and is conducted in two parts: instrument flight training and advanced contact flight training. During the instrument portion of training, the trainee learns to fly the UH-1 helicopter solely by reference to cockpit instruments in accordance with Instrument Flight Rules (IFR) as specified in Army Regulations 95-1.

Training is conducted in the 2C35 cockpit procedures trainer (4 hours), Device 2B24 (20 hours), and the UH-1 aircraft (30 hours). Advanced contact flight training consists of qualification (20 hours) and tactical flight training (45 hours) in the UH-1 aircraft. During Phase II training, trainees also receive 285 hours of academic instruction designed to support this phase of flight training. All Phase II training is conducted by military or Department of the Army civilian personnel.

A flow chart of current undergraduate flight training is shown in Figure 1.

The Experimental Training Program

In contrast to the current undergraduate flight training program described above, the experimental program proposes to replace the Primary Phase of training in the TH-55 helicopter with training in Device 2C35, Device 2B24, and the UH-1 aircraft. Students would receive training in

The current undergraduate flight training program described above has evolved since completion of the operational suitability test of Device 2B24. Major changes include (a) reduction of TH-55 training time in the Primary Phase from 110 hours to 85 hours, (b) replacement of 60 hours of TH-13T instrument training and 26 hours of instrument procedural training in the 1-CA-1 with 20 hours of instrument training in the 2B24 and 30 hours of instrument training in the UH-1, and (c) UH-1 advanced contact training reduced from 25 hours to 20 hours and tactical flight training increased from 25 hours to 45 hours in the UH-1. The earlier program is described in (1) and (2).

Figure 1 Current Undergraduate Flight Training Flow

EVASION ESCAPE &	24 25 26 27 28 29 30 31 32 33 34	CONTACT (4WKS) TACTICS (6 HKS) 20 HRS/UII-1 45 HRS/UII-1	NIGHT 15	NOE 15	Advanced Division
INSTRUMENT MAINT	12 13 14 15 16 17 18 19 20 21 22 23 24	SFTS (4 WKS) INST (6 WKS) 20 HRS 30 HRS/UH-1			Instrument Division
AIRMANSHIP AIRMANSHIP AIRMANSHIP	PF2 1 2 3 4 5 6 7 8 9 10 11 12 3	PRIMARY (12 WKS) 85 HRS/TH-55			Contractor
ACADEMICS	PF1*	THOITH 12			

*WOCs have 4 weeks preflight

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aircraft control techniques, UH-1 operating procedures, and instrument flight procedures in the 2C35 and the 2B24. Following this synthetic device training, students would transition to the UH-1 helicopter and be given sufficient aircraft training to meet Army student aviator performance requirements for entry into the Tactics portion of flight training.

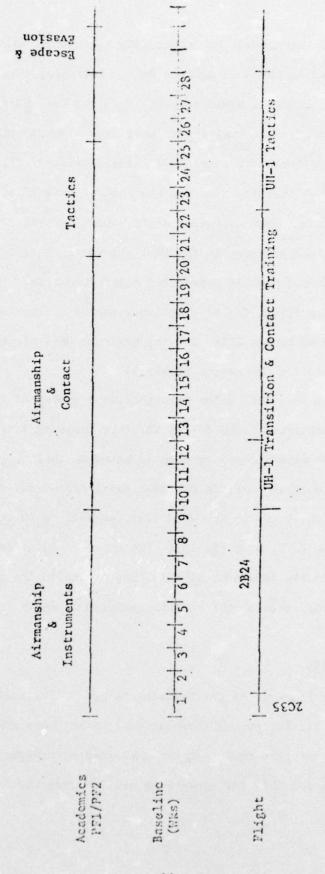
The amount of training time the student would require prior to entering Tactics has been estimated at 10 hours in Device 2C35, 75 hours in Device 2B24, and 75 hours in the UH-1 aircraft. A test of the proposed program should seek to answer two major questions: (a) how much of the undergraduate flight training syllabus can be accomplished through appropriate use of Device 2B24, and (b) how much UH-1 aircraft time would be required following the device training?

The program described below incorporates a number of training features and techniques found useful in other training programs. Their inclusion in the experimental program is based on their successful utilization, in whole or in part, in programs developed by the commercial airlines (13) and the U. S. Coast Guard (14), as well as in earlier Humro programs for the U. S. Army (5, 15). The major features and techniques are described in the remainder of this chapter and in the Appendices. A flow chart of the experimental training program is shown in Figure 2.

TRAINING CONCEPTS

Instructor Pilot as Flight Training Manager. Traditionally, instruction in flight simulators and training devices has been administered by non-rated enlisted personnel. In the experimental program, one instructor pilot is responsible for overseeing all training received by his

Figure 2 Experimental Undergraduate Flight Training Flow



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students—in the briefing room, the cockpit procedures trainer, the simulator, and the aircraft. It is believed that this concept provides a highly efficient training method which maximizes training continuity and instructor/student familiarity, simplifies performance recording, enhances flight safety, and allows individualized instruction. Also, since the instructor is aware of his students' knowledges, skills, and capabilities prior to proceeding to the next phase of training, unnecessary duplication of effort and loss of valuable training time are held to a minimum.

Functional Context Training/Crew or Peer Training. As much subject matter as possible is presented in a functional, or mission oriented, context, rather than in isolation. Emphasis is placed on how to operate the aircraft and its subsystems to accomplish a given mission. The configurations of the 2C35 and 2B24 are well-suited to this approach, since they allow for pilot and copilot training to be conducted simultaneously, with each performing his respective duties as he will later in operational missions. This concept also allows the students to assist each other in the solution of training problems.

Verbalization of Maneuver. The student is required to recite to the instructor how he is going to perform a flight maneuver before he attempts to perform it in the simulator or the aircraft for the first time. The use of this procedure is based on the assumption that if the student cannot verbalize the procedural and aircraft control components of the task, the chances are he will be unable to perform it.

<u>Proficiency Based Advancement</u>. Each student progresses through the program at his own pace. His rate of progress is based on his individual

learning ability and motivation, rather than on the requirement to complete any set number of hours of instruction. The introduction of new material is keyed to the attainment of objectively determined student performance on certain maneuvers, techniques, and procedures.

EXPERIMENTAL TRAINING MATERIALS

Course Outline

The traditional format for documenting a flight training program involves specification of maneuvers to be performed during each hour of scheduled training. This format is unsatisfactory for documenting an individualized course in which the rate of progress through the required exercises is a function of the trainee's rate of mastery of the maneuvers involved. Therefore, the traditional course documentation format is not used for this program, since it would not facilitate individualized training. Instead, a Course Outline (see Appendix B) was developed which lists all of the activities and maneuvers for which training is to be provided in the 2C35, the 2B24, and the UH-1, in the sequence in which they are to be introduced and initially practiced by each trainee.

Academics

In the traditional program, academic material is presented in a variety of ways: lectures, programed texts, conferences, demonstrations, exercises, etc. Primarily, programed texts are proposed for use in the experimental program, and the entire course is self-paced. Because of the method used to present initial flight training, sequencing of academic material will differ, in most instances, from that given the

traditional student. The only content changes from the current undergraduate flight training program in academics are related to material concerning the TH-55 helicopter. Since students in the experimental program are not required to fly the TH-55, there is no requirement for them to learn systems information covering that aircraft. In Appendix C, academic subjects have been arranged in a proposed sequence which parallels the instruction the student will receive in the 2C35, the 2B24, and the UH-1. Existing Aviation Center programed texts have been utilized wherever possible. Where programed texts are not available, proposed alternate media are indicated. The sequence shown is an approximation. Undoubtedly, changes would occur based on first-use experience. However, in order to assure that the student is adequately prepared academically for the instruction he will receive in the devices and the aircraft, he should have completed Items 1 through 6 (in Appendix C) prior to beginning training in the 2B24; Items 7 through 57 prior to beginning training in the UH-1 aircraft; and Items 58 through 90 prior to starting tactical training.

Data Forms

The data forms appearing in Appendices D, E, and F were developed for use during a planned evaluation of the proposed new program. As such, they are not necessarily considered as being suitable for operational use without testing and, probably modification. They are

¹It should be noted that a complete systems engineering of academic instruction in undergraduate flight training was beyond the scope of the present effort. It is believed, however, that substantial improvements to the proposed program could be achieved through such an undertaking.

presented here to provide examples of the type of information deemed necessary in a program of this sort.

Since most instructors will be unfamiliar with the use of these forms, considerable method of instruction (MOI) training, in addition to that typically given instructor pilots, will be required before instructors can be expected to employ the forms effectively.

Maneuver Performance Record Forms (MPRFs). The MPRF was designed to serve three principal functions, and reference to the specimen set in Appendix D will illustrate each. The first is to provide the instructor with an objective record form in which he can represent graphically the daily progress of each student. This purpose is achieved by requiring that he use the MPRF to score the student's first performance of a maneuver each training period. A (\checkmark) or an (x) is used to indicate satisfactory or unsatisfactory performance, respectively, of each maneuver element.

The second function of the MPRF is to provide performance—
specific feedback to the trainee concerning his performance on each
maneuver he practiced each day, and to enable him to evaluate his
progress with respect to that maneuver by comparing that day's performance with previous days' performance. The MPRF thus becomes a very
effective communication tool between trainee and instructor.

The third function of the MPRF is to limit the amount of training the trainee receives in each maneuver. By requiring instructors to move on to other items as soon as the criterion of one errorless trial is achieved in a particular maneuver, the tendency of some instructors to overtrain can be reduced.

Amplified Checklist. The Amplified Checklist, a copy of which is at Appendix E, has been adapted from the descriptions of cockpit procedures given in the UH-1H Operator's Manual and is used as described in the Outline of 2C35 Training Periods shown in Appendix H. The checklist describes the tasks and sequences to be followed in the performance of various cockpit procedures. Included in the checklist is information concerning aircraft systems operation in both normal and abnormal modes. It should be noted that this underlying systmems information is presented in the context of actually performing the procedure. Space is provided on the checklist for recording student errors. Use of the checklist is continued until the criterion of one errorless trial is reached, and it thus functions similarly to the MPRF.

Checkride Forms. The checkride forms shown in Appendix F have been prepared to provide a standardized tool for determining student performance at various points in training. The checkride forms are to be completed by flight examiners who have themselves been trained in the proper administration of the checkride. These forms are more detailed than those typically used in Army checkride evaluations, but they serve the same function. Forms are included for a pre-solo evaluation, an instrument checkride, and an advanced contact checkride. Additional checkrides for intermediate stages of training could, of course, be designed.

An exception is the Army's use of the Pilot Performance Description Record (PPDR) in the Primary Phase of undergraduate flight training. The PPDR is a product of previous HumRRO research.

In the traditional flight training program, learning to solo the helicopter is the first major hurdle facing the student. Students are expected to solo within their first 15 flight hours, and failure to do so within 20 hours may result in elimination from the flight training program. By contrast, students in the experimental program will not only have "soloed" the 2B24, but also will have developed flight skills far in advance of those of the typical pre-solo stage trainee. Consequently, there is no requirement in the new program to solo the UH-1 within a specified number of hours. In fact, solo is intended to be delayed so that greater proficiency can be built over a variety of skill areas. It is required, however, that the student demonstrate satisfactory performance of certain maneuvers and procedures prior to being allowed to solo. These maneuvers and procedures are indicated on the pre-solo data forms shown in Appendix F.

Most flight instructors would be generally familiar with the contents of the forms provided for the instrument checkride and the advanced contact checkride, since the performance standards for these do not differ from those employed in the usual undergraduate flight training program.

Mission Model

The outline for a basic IFR mission found at Appendix G has been developed as a general model of a typical IFR mission and represents the context within which all instrument training subsequent to that described in Appendix I is to be given. Daily flight training periods would consist of practice of all or portions of the listed items.

ADMINISTRATIVE PROCEDURES

General

In all phases of training—in the procedures trainer, the simulator, and the aircraft—each instructor will have two students. It is anticipated that each instructor will have the same two students throughout training.

Instructional periods will be two hours in Device 2C35 and three hours in Device 2B24. As nearly as possible, each period will be divided so that each student will have an equal opportunity to perform both pilot and copilot duties. Instructional periods in the UH-1 aircraft will remain as they are in the current program, i.e., of three hours duration, with each student spending approximately half that time acting as pilot and the other half observing the other student perform the pilot's duties (the instructor acts as copilot).

The typical flight instructor will be unfamiliar with teaching basic flight maneuvers to flight naive students in Device 2B24. He will, therefore, require some additional MOI training prior to teaching in the experimental program.

Device 2C35 Training

To use the 2B24 to provide the student with cockpit orientation, instrument familiarization, and initial systems training would be economically unwise. Therefore, a program to provide this initial training was designed around the 2C35 cockpit procedures trainer. The objectives of this training are to:

(a) Familiarize the student with the UH-1 cockpit environment;

- (b) Teach him the location of all cockpit instruments, their purposes, and how to read them;
- (c) Teach him how to use the pilot's checklist (TM 55-1520-210-10CL);
- (d) Teach him the systems information necessary to insure that he can perform safe aircraft starts, run-ups, and shutdowns;
- (e) Teach him to perform the pre-start check, engine start, engine run-up, engine shutdown, and how to use Forms 2408-12 and -13.
- (f) Teach him the appropriate procedures for the following emergencies: hot start, hung start, low battery voltage, abnormal oil pressure, and high and low side governor failure.

In order to accomplish the above objectives, the student must learn:

- (a) to identify the checklist item;
- (b) the location of the item;
- (c) to perform the task associated with the item (what to do with it and what happens);
- (d) the required systems information which will allow him to use the system efficiently and to respond to cues indicating system or component parts malfunction and to take appropriate action to insure safety of flight.

Each student will be issued programed texts PT 55, Engine Start and Run-up, UH-1D/H, and PT 2018A, Description of UH-1H, which he will complete in that sequence prior to reporting to the 2C35 for the first period of instruction.

Each student will also be furnished copies of the pilot's checklist, TM 55-1520-210-10CL, and of the Amplified Checklist (from TM 55-1520-210-10, UH-1D/H Operator's Manual) shown in Appendix E.

It is intended that the instructors be trained during MOI to administer the training following the Course Outline (Appendix B) in the manner indicated in Appendix H (Outline of 2C35 Training Periods). The instructor initially assumes the teacher's role, explaining procedures and locating knobs, levers, and switches and defining their purposes and operation. As the student becomes more proficient, the instructor's role changes to that of moderator, evaluator, and interrogator. He initiates peer instruction, interacting only when the student can't solve the problem; scores errors; questions the students about instructional material covered; inserts malfunctions into the system to test knowledge of procedures, etc.

Speed in accomplishing item tasks on the checklist is desirable, but accuracy has priority. Operationally, cockpit procedures must be accomplished in periods of darkness, when cockpit visibility is degraded, as well as in daylight. Therefore, any technique which might insure that an item task is not improperly accomplished, or accidentally eliminated, should be used.

During the preliminary work with the experimental training program, the 2C35 training described here was thoroughly tested and found to be very effective in achieving the 2C35 training objectives indicated above. While it is anticipated that most students will master the objectives within approximately five 2C35 periods, they should be encouraged to practice using the checklist on their own time, either in a vacant 2C35

or at the learning center. Appendix H contains a detailed outline of the activities to take place during 2C35 training periods. Students who reach criterion performance are scheduled to begin Device 2B24 training the next period.

Device 2B24 Training

The purpose of this training is to teach the flight naive student the knowledges and skills required to enable him to operate the UH-1 under both normal and abnormal conditions and to complete successfully a standard instrument checkride. Students in the program, unlike traditionally trained students, will have had little or no flight training. While some students may have had ROTC fixed wing training, many may possess only those aviator knowledges and skills learned in the 2C35. The task for the instructor in this program is to take an individual who has never flown a helicopter before and train him to Army aviator instrument proficiency standards solely in Device 2B24.

The flight instructor will normally start the 2B24 training program in the copilot's seat. However, as quickly as possible, he will relinquish this position to his other student and initiate simultaneous crew training, at which point he will occupy the instructor/observer seat. From this position, using on-board controls and with assistance from the console operator, the instructor is able to control the training activities as well as the training environment. The 2B24 incorporates many unique capabilities designed to enhance the conduct of training. These capabilities are described in documents provided by the device manufacturer, and the instructor must be trained to use them effectively during MOI.

Training should be conducted in the manner indicated in Appendix I (Device 2B24 Training Periods) until the student has attained sufficient proficiency to pass a standard instrument checkride in the device. That checkride should be administered by an independent examiner (not by the student's instructor or another instructor from the same training flight) to assure objective evaluation of student capabilities (16).

UH-1 Aircraft Training

Following the successful completion of a standard instrument checkride in Device 2B24, students begin training in the UH-1 aircraft. The
purpose of this training is threefold: (a) validation of instrument
training received in the 2B24; (b) VFR qualification in the UH-1 aircraft; and (c) acquisition of the requisite skills and knowledges for
entry into the Tactics Phase of UH-1 flight training.

To take full advantage of the positive training transfer from the 2B24 to the UH-1 aircraft and to familiarize the student with the aircraft and visual flight environment, the first flight should be flown along a route practiced in the 2B24, using both visual and instrument flight cues. The first part of the flight should be conducted under the hood, with the hood being removed occasionally to allow the student to compare visual and instrument cues. Visual flight training will be conducted using the integrated concept where practicable, and should follow the sequence outline in Appendix B (a more detailed outline for training periods is not provided since flight instructors are already familiar with instructing in the aircraft). As a supplemental aid to the maintenance of instrument flight proficiency during visual flight training, occasional 2B24 training periods will be scheduled.

Training continues in the fashion indicated until the student can pass an advanced contact checkride, at which time he advances to the Tactics Phase of flight training.

As noted earlier, the training program described in this report has not been tested. Portions of the program were tried out during the developmental process, but a planned full-scale evaluation of the program's feasibility for Army use was not accomplished. It is believed that the material in this report would be very useful if such an evaluation were undertaken.

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APPENDICES

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APPENDIX A

PUBLICATIONS AND RESEARCH BY-PRODUCTS OF WORK UNIT SYNTRAIN

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Research By-Products

U-21 Cockpit Procedures Trainer

* U-21 CPT Program of Instruction

U-21 Paper Trainer

T-42 Paper Trainer

OV-1 Paper Trainer

GAT-2 Checklist

- * Program of Instruction for Twin-Engine Qualification and Instrument Training
- * Program of Instruction for Combat Readiness Proficiency Training
- * Program of Instruction for Rotary Wing Instrument Training
- * Program of Instruction for Initial Entry Rotary Wing Training

 IFR Typical Mission Task List

* Method of Instruction (MOI) material included

APPENDIX B

COURSE OUTLINE

A, 2C35 Cockpit Procedures Training

- 1. Pre-start check
- 2. Start
- 3. Run-up
- 4. Emergencies
 - a. Hot start
 - b. Hung start
 - c. Low battery voltage
 - d. Abnormal engine oil pressure
 - e. No governor failure

B. 2B24 Flight Training

- 1. 2B24 briefing
 - a. Description
 - b. Capabilities
- 2. Cockpit crientation
 - a. Door-motion interlock
 - b. Motion system and control
 - c. Freeze function and control
 - d. Continue flight function and control
 - e. Overhead light and control
 - f. Cathode Ray Tube functions
 - g. TV camera (if applicable)
 - h. Instructor station

3. Development of flight control skills (IFR references)

- a. Roll control
 - (1) Attitude indicator (roll component)
 - (2) Lateral cyclic control
 - (3) Radio magnetic indicator (RMI)
- b. Pitch control
 - (1) Attitude indicator (pitch component)
 - (2) Fore and aft cyclic control
 - (3) Airspeed indicator
 - (4) Altimeter
 - (5) vertical velocity indicator

Power control c. (1) Collective pitch control (2) Torque pressure gauge (3) Effect on trim and use of pedals (4) Effect on airspeed (5) Effect on altitude (6) Use in turns (7) Effect of altitude on power Cruise flight (1) Maintaining heading (a) RMI (b) Lateral control (c) Trim (2) Maintaining altitude (a) Altimeter (b) Pitch control (c) Airspeed indicator (d) Vertical velocity indicator Changing airspeed e. (1) Power (2) Attitude changes (3) Heading control (4) Trim Changing heading, standard and half-standard rate (1) Roll-in rate (2) Angle of bank (3) Turn needle (4) Altitude control (5) Airspeed control (6) Trim (7) Roll-out rate Changing altitude Power (1) (a) Torque Meter (b) Vertical velocity indicator (2) Pitch control (a) Attitude indicator (b) Airspeed (3) Trim (4) Heading control (a) RMI (b) Attitude indicator (5) Altimeter Level off (1) Altimeter (2) Power (3) Pitch (4) Trim (5) RMI

Changing heading and altitude

(1) Entry

- (a) Power
- (b) Angle of bank
- (c) Roll-in rate
- (d) Pitch attitude
- (e) Trim
- (2) Maintaining
 - (a) Turn needle
 - (b) Vertical velocity indicator
 - (c) Airspeed
 - (d) Power
- (3) Roll-out
 - (a) RMI
 - (b) Attitude indicator
- (4) Level off
 - (a) Altimeter
 - (b) Power
 - (c) Pitch
 - (d) Trim
 - (e) RMI
- Instrument takeoff (Demonstrate)
 - (1) Pre-takeoff check
 - (2) Power application
 - (3) Attitude control
 - (a) Pitch
 - (b) Ro11
 - (4) Light on skids
 - (5) Engine instruments check
 - (6) Heading control
 - (7) Transition to climb, 500 FPM (Demonstrate)
 - (a) Power
 - (b) Vertical velocity indicator
 - (c) Airspeed
 - (d) Pitch
 - (e) Altimeter
 - (f) Trim
 - (8) Level off (Demonstrate)
 - (a) Power
 - (b) Pitch
 - (c) Trim
 - (d) Level off check
- k. Ground controlled approach
 - (1) Radio communication procedure
 - (2) ASR
 - (3) PAR
- 1. Unusual attitudes
 - (1) Roll attitude
 - (2) Pitch attitude
 - (3) Power
 - (4) Trim
- Radio procedures, IFR navigation, and emergency procedures (Demonstrate and Practice)
 - a. VOR

- (1) Orientation
- (2) Tracking
- (3) Station passage
- (4) Reporting
- (5) Position fixing
- (6) Holding procedures
- (7) VOR approach
- (8) VOR missed approach
- (9) Cross-country flight planning
- Under- and overspeeding N₂ governor failure (Demonstrate and Practice)
- Engine failure emergency procedures--power recovery, IFR (Demonstrate and Practice)
- d. ADF (Demonstrate and Practice)
 - (1) Orientation
 - (2) Tracking
 - (3) Station passage
 - (4) Position fixing
 - (5) Holding procedures
 - (6) ADF approach
 - (7) ADF missed approach
- e. Radar procedures (Demonstrate and Practice)
- f. Two-way radio failure
- g. ILS (Demonstrate and Practice)
 - (1) Localizer interception
 - (a) Radar vector
 - (b) VOR radial
 - (c) ADF bearing
 - (2) Outer marker
 - (3) Localizer tracking
 - (4) Glide slope following
 - (5) Approach
 - (6) Missed approach
- 5. Cross-country exercises
 Cross-country exercises should be planned to provide the student with an opportunity to practice all the skills, techniques, and procedures required for a standard instrument rating. These exercises should be practiced until that level of proficiency is reached. As soon as the student can demonstrate reasonable control of the device, as much training as possible should be conducted in the cross-country context.
- C. UH-1 Contact Training and Instrument Flight Practice
 - 1. Pre-flight and post-flight procedures

- 2. Cross-country flight
- 3. Hovering flight
 - a. Stationary hover
 - b. Landing from a hover
 - c. Takeoff to a hover
 - d. Hovering turns
 - e. Clearing turns (90° turns)
 - f. Anti-torque failure while hovering
 - g. Moving hover
 - h. Hovering autorotation
- 4. Deceleration
- 5. Use of forced trim
- 6. Traffic pattern
 - a. Normal takeoff
 - b. Upwind leg
 - c. Crosswind legd. Downwind leg

 - e. Normal approach
 - f. Exiting traffic pattern (leaving traffic)
 - g. Entering traffic pattern (entering traffic)
 - h. Go-around procedures
- 7. Servo-off flight
- 8. Autorotations
 - a. Straight-in
 - b. With turn
- 9. Forced landings
- 10. Power recovery
- 11. Pre-solo checkride
- 12. Instrument checkride
- 13. Stagefield proficiency
 - a. Review and practice to required proficiency all previous maneuvers
 - b. 180° hovering turns
 - c. 360° hovering turns
 - d. Quick stops .
 - e. Emergency procedures
 - f. Internal loads
 - g. Use of auxiliary equipment
 - h. Hydraulic-off landing

- i. Simulated anti-torque failure
- j. Maximum performance takeoff
- k. Steep approaches
- 14. Confined area, pinnacle, slope operations, and night flight
 - a. Confined area and pinnacle operations
 - (1) High reconnaissance
 - (2) Low reconnaissance
 - (3) Approach
 - (4) Ground reconnaissance
 - b. Slope operations
 - c. Night flight
 - d. Advanced contact checkride
- 15. Cross-country navigation
 - a. Flight planning using aeronautical and tactical maps
 - (1) Route plotting
 - (2) Computing distance and direction
 - (3) Course determination
 - (4) Checkpoint selection
 - (5) Off-course corrections

APPENDIX C

ACADEMICS

Tit	<u>le</u>	Type Instruction	k
1.	History and Development of the Helico and Aircraft Orientation	opter 1.0 PI	
2.	Rotary Wing Aerodynamics	0.6 PI 0.4 TV	
3.	Physical Laws	1.0 PI	
4.	General Description UH-1H	2.0 PI	
5.	Engine Start and Runup UH-1H/D	2.0 PI	
6.	Flight Instruments	3.0 PI	
7.	Flight Controls	3.0 C	
8.	General and Visual Flight Rules	3.0 PI	
9.	Pitot Static System and Magnetic Comp	pass 2.0 PI	
10.	DOD FLIP	2.0 PE 4.0 PI	
11.	Technical Publications, Aircraft Form and Records, and Inspections	ms 2.0 PI	
12.	Communication Procedures and Radio Phraseology	1.0 PI	
13.	Radar Procedures	1.7 PI 0.3 TV	
14.	PAR Approach	0.7 PI 0.3 TV	
15.	VOR Orientation and Tracking	1.6 PI 0.4 TV	
16.	VOR Enroute Navigation	2.0 PI 0.5 TV	
C		GS = Guest Speaker	

- D = Demonstration
- E = Examination
- F = Film
- FEX = Field Exercise

- PE = Practical Exercise
- PI = Programed Instruction
- TV = Television

Title	<u>e</u>	Type Ins	truction		
17.	Instrument Flight Rules	2.0 PI			
18.	VOR Approach Procedures	2.0 PI			
19.	Transponder	1.0 PI			
20.	Holding Procedures	2.0 PI			
21.	Navigational Computer Slide Rule	3.0 PI			
22.	Navigational Computer Wind Face	2.0 PI			
23.	ADF Navigation	2.3 PI	0.7 TV		
24.	ADF Approach	1.0 PI			
25.	Enroute Charts and Supplement	2.5 PI			
26.	ATC Agencies	3.0 PI			
27.	IFR Communications	4.0 PI			
28.	ATC Clearances	2.0 PI			
29.	ILS	1.8 PI	0.7 TV		
30.	The Atmosphere	1.5 PI	0.5 F		
31.	Pressure and Winds	1.5 PI	0.5 F		
32.	Air Masses and Clouds	0.6 PI	0.4 TV		
33.	Frontal Weather	2.0 PI	1.0 E		
34.	Thunderstorms	1.0 PI			
35.	Sequence Reports, Forecasts and Winds Aloft	4.6 PI	0.4 TV		k
36.	Surface Weather Map	1.0 PI			
37.	Weather Depiction and Radar Summary Charts	1.0 PI			
38.	Weather Hazards	0.7 PI	1.5 F	0.8	PE
39.	IFR Flight Plans	2.3 PI	3.0 PE	0.7	TV
40.	Communications Failure and Emergency Procedures	4.0 PI			

Titl	<u>e</u>	Type Ins	truction
41.	Weight and Balance	1.0 PI	1.0 PE
42.	Gyrocompass Failure	1.0 PI	
43.	Weather Fundamentals and Flight Planning	2.0 PI	
44.	IFR Flight Planning	14.3 PE	0.7 TV
45.	Aviation Fuels and Lubricants	1.0 PI	
46.	Aircraft Hardware and Safetying Procedures	1.0 PI	
47.	Fort Rucker Downed Aircraft Procedures	0.7 PI	1.3 TV
48.	Administrative Responsibilities	1.0 PI	
49.	Aircraft Accident Prevention and Investigation	1.0 PI	
50.	Aeronautical Chart Symbols	2.0 PI	
51.	Distance and Direction	0.7 PE	1.3 PI
52.	Variation, Deviation and Wind Effect	2.0 PI	
53.	Navigational Practical Exercise	1.0 PE	
54.	Off Course Corrections	1.0 PI	
55.	Performance Data Charts	1.0 PI	
56.	Navigation Practical Exercise II and III	2.0 PE	
57.	Aviation Medicine Orientation	2.0 C	
58.	Wake Turbulence	0.7 PI	0.3 TV
59.	Proximity Warning Device	1.0 PI	
60.	Flight Plans (VFR)	1.0 PI	
61.	Preflight and Postflight Inspections	2.0 PI	1.0 PE
62.	Flight Line Orientation	2.0 C	
63.	Forces Acting on the Rotor System	1.7 C	0.3 F
64.	Rotor System in Translation	2.4 C	0.6 F
65.	Rotor Blade Stalls	0.7 C	0.3 F

Title	2	Type Ins	truction
66.	Use of Operator's Manual	1.0 PI	1.0 PE
67.	Electrical System Components and Malfunctions	1.0 PI	1.0 PE
68.	Power Train Components	1.0 PI	1.0 C
69.	Main and Tail Rotor Systems	1.0 PI	1.0 C
70.	Fuel System	1.0 PI	1.0 C
71.	Engine Failure Emergency Procedure	3.0 D 1.7 C	1.0 P1 0.3 TV
72.	Precautionary Measures and Emergency Procedures	0.3 TV	2.7 PI
73.	Malfunction and Analysis	1.0 PI	1.0 C
74.	Accident Prevention Literature	1.0 C	
75.	Unit Training and Standardization	1.0 C	
76.	Aircraft Operations and Emergency Procedures	1.0 C	
77.	Examination (Safety)	1.0 E	
78.	Altitude Physiology		2.0 PE1 10 hours de chamber)
79.	G-Forces	1.0 C	
80.	Toxic Hazards in Aviation	1.0 C	
81.	Adverse Effects of Temperature Extremes	1.0 C	
82.	Noise in Aviation	1.0 C	
83.	Protective Equipment	1.0 C	
84.	Night Vision Orientation	0.5 C	0.5 D
85.	Spatial Disorientation and Sensory Illusions of Flight	1.0 C	
86.	Stress and Fatigue	1.0 C	
87.	Review of Aviation Medicine and Life Support	1.0 C	

Titl	<u>e</u>	Type Ins	truction	
88.	Examination	1.0 E		
89.	Tactical Academic Orientation	1.0 C		
90.	Introduction to NATO Ground and Air Forces	2.0 GS		
		0.5 C	1.5 F	
91.	Threat	2.5 C	0.5 F	
92.	Tactical Aircraft Communications	2.0 C		
93.	Unattended Ground Sensors	0.3	0.7 F	
94.	Aircraft Flares	1.3 C	0.5 D	0.2 F
95.	Combat Intelligence	1.0 C		
96.	Helicopter Tactical Loads	2.0 PI		
97.	Communications Security	2.5 C	0.5 D	
98.	Communications Procedures	0.5 C	2.5 PE	
99.	Electronic Warfare	2.0 C		
100.	Air Cavalry Operations	2.0 C		
101.	Examination	2.0 E		
102.	Offense/Defense	3.0 C		
103.	Aerial Observation	2.0 C		
104.	Law of Land Warfare	1.0 C		
105.	Airmobile Operations	1.5 C	0.5 F	1.0 PE
106.	Aircraft Vulnerability to Special Weapons	1.0 C		
107.	Tactical Flight NOE Briefing	2.0 C		
108.	Target Identification	2.0 C	2.0 PE3	
109.	Pathfinders and Tactical Landing Areas	1.0 C		
110.	Examination	2.0 E		
111.	Tactical Navigation	7.3 C	7.0 PE3	0.7 F

Title		Type Ins	truction	
112.	Examination	3.0 E		
113.	Fire Support Coordination	0.5 C	0.5 F	
114.	Aerial Adjustment of Artillery	4.0 C	8.0 PE3	
115.	Attack Helicopter Missions and Employment	2.0 C		
116.	Examination	3.0 E		
117.	Introduction to Survival	1.5 C		
118.	Survival Medicine	2.0 C		
119.	Overwater Survival	0.7 C	0.3 D	
120.	Physiology of Food	1.0 TV		
121.	Travel Techniques	1.5 C		
122.	Camouflage	2.0 C	1.0 TV	1.0 PE1
123.	Evasion	1.0 C		
124.	Introduction to Resistance	2.0 C		
125.	PW Organization	1.5 C		
126.	Field Exercise	24.0 FEX		
127.	Examination	2.0 E		
128.	Prisoner Exploitation	3.0 C	2.0 F	1.0 TV

APPENDIX D

MANEUVER PERFORMANCE RECORD FORMS

MPRF - 1

			AND LEVEL	С	ruise	
STUDENT					Sheet No). <u> </u>
Date						
Altitude	(±100')					
Direction	(±10°)					
Ball cent	ered (th ball)					Г
	Errors					
	Total prior time:	1st trial		Criteria	reached	

	LEVEL TURNS Cruise (Standard rate) 90°
Student	Sheet No.
Date	
Roll-in rate 53	
Rate of turn (*, needle)	
Altitude (±100°)	
Degree turned (:10°)	
Roll-out rate	
Ball centered (£1, ball)	
Errors	
Total prior time	1st trial : Criteria reached :

19

Date ACCELERATION Power increase: 4 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	Date ACCELERATION Power increase: 4 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±1/2 ball) Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±1/2 ball) Heading control ±10° Airspeed: 70 kts (±5 kts)		ERATION AND	DECET LPAT.	UN .		
ACCELERATION Power increase: 4 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	ACCELERATION Power increase: 4 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 100 kts (±5 kts) DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Student				Sheet No.	-
Pitch attitude: adjust to maintain alt. (±50') Ball centered (±12 ball) Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±15 ball)	Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±½ ball) Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Date	ACCELERA	TION			
tain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	tain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Power increase: 4 lbs (±1 lb)					
Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	Heading control ±10° Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Pitch attitude: adjust to maintain alt. (±50')					
Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	Airspeed: 100 kts (±5 kts) Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Ball centered (±12 ball)					
Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 lb) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±1, ball)	Power: reduce 2 lbs DECELERATION Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50°) Ball centered (±½ bail) Heading control ±10° Airspeed: 70 kts (±5 kts)	Heading control ±10°					
DECELERATION Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Airspeed: 100 kts (#5 kts)					
Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball)	Power decrease: 8 lbs (±1 ib) Pitch attitude: adjust to maintain alt. (±50') Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Power: reduce 2 lbs	DECELERA	TION			
tain alt. (±50') Ball centered (±1, ball)	tain alt. (±50°) Ball centered (±½ ball) Heading control ±10° Airspeed: 70 kts (±5 kts)	Power decrease: 8 lbs (±1 lb)					
	Heading control ±10° Airspeed: 70 kts (±5 kts)	Pitch attitude: adjust to maintain alt. (±50')					
Heading control ±10°	Airspeed: 70 kts (±5 kts)	Ball centered (±1, ball)					
toward because because the	Inches I have been been been been been been been be	Heading control ±10°					
Nirspeed: 70 kts (±S kts)	Power increase: 2 the	Airspeed: 70 kts (±5 kts)					
International International International International International	roser merease. 2 108	Power increase: 2 lbs					

	RATION AND I				
ACCEL Power increase: 6 lbs (±1 lb)	ERATION TO I	VORMAL CRUTS) E		
Pitch attitude: adjust to maintain alt. (250')					
Ball centered (\mathfrak{t}_2 ball)					
Heading control ±10°					
Airspeed: 90 kts (±5 kts)					
Power decrease: 2 lbs					
Errors					
Total prior time:	lst trial		Criteria	reached	

ALIL	CLIMB (90				
(Record for 2000' from Student			o start of	level off) Sheet No	
Date					
Power increase 5 lbs (*1 lb					
Attitude: level (t) bar)					
Vertical speed 500 FPM (*160°)					
Airspeed 90 kts (=10 kts)					
Ball centered (th ball)					
Direction VOR ±5°, ADF ±5°, Hdg ±10°					
Led altitude (40°)	LEVEL OFF	90 KTS			_
Airspeed 90 kts (±10 kts)					
Power reduced 5 lbs (±1 lb)					
Altitude (±100')					
Direction VOR ±5°, ADF ±5°, Hdg ±10°					
Ball centered (t) ball)					
Errers	r1				
Total prior time:	1-t trial		criteri	reached	

	DESCENT (S	90 KTS)			
(Record for 2000' from)	initial pitch	change to	start of 16	Sheet No.	
Date					
Power decrease 5 lbs (±1 lb)					
Attitude level (±12 bar)					
Airspeed 90 kts (±10 kts)					
Ball centered (±1/2 ball)					
Direction VOR ±5°, ADF ±5°, Hdg ±10°					
	LEVEL OFF	90 KTS			
Led altitude (40')					L
Airspeed 90 kts (±10 kts)					
Power increased 5 lbs (±1 lb)					
Attitude (±100')					
Direction VOR ±5°, ADF ±5°, Hdg ±10°					
Ball centered (±4 ball)					<u></u>
Errors					
Total prior time:	1st trial		Criteria	reached	[:

	INSTRUMENT	TAKE-OFF			
Student				Sheet	No
Date					
Light on skids	<u> </u>				
Power increase 5 lbs.(±1 lb.)					
Heading (10°)					
Attitude level (±4 bar)					
Established positive climb (Altimeter moving clockwise)					
Attitude: 1 bar low (-0, +1 bar)					
No assistance required					
Heading to 200' (±10°)					
INITIAL CLIMB (ST Attitude: 80 kts (±10 kts)	RAIGHT) 20	000' AFTER 80	KTS ESTABL	ISHED	
Accidade, 60 kts (210 kts)	L	J		-	
Power 500 FPM (±100FPM)					
Heading (±10°)					
Trim proper (±½ ball)					
Errors	<u></u>				
Total prior time:	1st tri	ial [Criteria	reached	

CHAN	GING ALTITUDE AND HEADING CLIMBING TURNS (90 KTS - 500 FPM)	
Student	STANDARD RATE	Sheet No
Pate	ENTRY	
Power increase 5 lbs (±1 lb)		
Pitch, bank and power control	MAINTAINING	
Airspeed 90 kts (±10 kts)		
Rate of turn (standard rate the needle)		
Vertical speed 500 FPM (±100')		
Ball centered (±1/2 ball)	RECOVERY 90 KTS	
Led altitude (40')		
Pitch, bank and power control		
Degrees turned (±10°)		
Altitude 500' (±50')		
Power reduced 5 lbs (±1 lb)		
Ball centered (±1, ball)		
Airspeed 90 kts (±10 kts)		
Errors		
Total prior time:	1st trial : Criteri	a reached :

CHANG	DESCEND (90 KTS	DE AND HEADI ING TURNS - 500 FPM) ARD RATE	NG		
Student				Sheet No.	_
Date	ENT				Г
Power decrease 5 lbs (±1 lb)	L. C.N.				
Pitch, bank and power control	MAINTAL				
Airspeed 90 kts (±10 kts)	FEGRIA			$\Gamma^{}$	Г
Rate of turn (standard rate ±1, needle)					
Vertical speed 500 FPM (±100')					
Ball centered (±% bull)	RECO				
Led altitude (46°)					
Pitch, bank and power control					
Degrees turned (±10°)					
Altitude 500' (±50')					
Power increased 5 lbs (±1 lb)					
Ball centered (±1 ball)					
Airspeed 90 kts (±10 kts)					
Errors					
Total prior time:	lst tria		Criteria 1	reached	Г

-

ě.	GCA	
Student		Sheet No.
Date	INITIAL CALL	
Position report	TRANSITION	
Transponder procedures	INANSTITON	
Heading control (±10°)		
Altitude (±100')		
Airspeed (±10 kts)		
Rolled out on heading (±10°)	DOWNWIND	
Heading control (±10°)		
Altitude (±100')		
Airspeed (±10 kts)		
Pre-landing check		
Rolled out on heading (±10°)	BASE LEG	
Heading control (±10°)		
Altitude (±100')		
Airspeed (±10 kts)		
Final cockpit check		

	GG	CA				
	FI	NAL				
Rolled out on heading (±10°)						
Heading control (±10°)						_
Altitude (±100')						
Heading (±5°)	APPI	ROACH				
Glideslope (follows instructions)						_
Voice procedures] [] [_
Errors						_
Total prior time:	1st tria	al :	Criter	ia reac	hed	

VOR ORT	ENTATION & RADIAL INTERCEPTION Sheet No
Date	
Tuned station properly	
Identified station	
Identified radial	
Turned to proper intercept heading (±10°)	
Rolled out on course (±5°)	
Altitude (±100')	
Errors	
Total prior time:	1st trial : Criteria reached :

	TRACK FOLLOWING (Record for five minutes)
Student	Sheet No.
Date	
Maintained track (±5°)	
Altitude (±100')	
Errors	
Total prior time:	1st trial : Criteria reached

UNUSUAL ATTITUDES Steep Descending Turn)		
	Sheet N	0
RECOVERY		
lst trial :	Criteria reached	:
	RECOVERY	Sheet N

		INDICIAL APPLICACIONES		
		UNUSUAL ATTITUDES (Steep Climbing Turn)		
Student			Sheet No	
Date				
Wings:	level attitude	RECOVERY		
Pitch:	level attitude			
Power:	adjust			
Trin				
	Errors			
	Total prior time:	1st trial :	Criteria reached	

	RADIO CROSS CO	OUNTRY VOR			
C1		AUT		C) V	
Student				Sheet No.	
Date	PREPARATION	FOR TAKE OFF			
Radio calls					
Clearance					
Transponder	TAKE OFF & CLIN	MR TO ALTIT	DE.		
Track (±5°) ADF ±5°					
Reports					
Procedures	ENRO	DUTE			
1st leg					
Altitude (±100')					
Track (±5°) ADF ±5°					
Ground speed computed					
Reports					
ETA					
Procedures					

	PURIO CROSS COUNTRY VOR
2nd leg	RADIO CROSS COUNTRY ADF
Altitude (±100')	
Track (±5°) ADF ±5°	
Track (±5") ADF ±5"	
Ground speed computed	
Reports	
ETA	
Procedures	
3rd leg	
Altitude (±100')	
Track (±5°) ADF ±5°	
Ground speed computed	
Reports	
ETA	
Procedures	

	RADIO CROSS COUNTRY VOR		
4th leg			
Altitude (±100')			
Track(±5°) ADF ±5°			
Ground Speed Computed			
Reports			
ETA			
Procedures			
ETA	DESTINATION		
Report			
Holding			
Pre-landing check complete	APPROACH		
Time outbound			
Procedure turn altitude			
Track (±5°) ADF ±5°			
Low station altitude			
Min. altitude (-0, +100')			
Airspeed (90 kts ±10 kts)			
Reports			

		RADIO CROSS	ADI		
Time		MISSED	APPROACH		
Power		ļ			
Report					
Procedure					
	Errors				
	Total prior time:	1st tri	al j	Criteria reach	ned j

	ENGINE FAILURE (During Flight)	
Student	Sheet	No.
Date		
Rotor RPM maintained (collective down)		
Ball centered (±1/2 ball)		
Glide established 80 kts (±10 kts)		
Called out RPM/N ₁ /AS		
Heading maintained (±10°)		
If time allows: Transponder set to EMER		
Transmit MAYDAY (3 times)		
Transmit A/C identification		
Transmit position		
Transmit nature of distress		
Main fuel: OFF (verbal only)		
Battery switch: OFF (verbal only)		
Shoulder harness: LOCK (verbal only)		
Errors		
Total prior time:	1st trial : Criteria reached	1

H	DRAULIC PO	WER FAILURE			
Student				Sheet No.	
Date					
Adjust airspeed					
Hydraulic control circuit breaker: OUT					
Hyd. control circuit breaker: IN (if power not restored)					
Hydraulic control switch: OFF					
Master caution warning light: RESET					
Landing: Accomplish					
Errors:					
Total prior time:	lst trial		Criteria	reached	
					•

OVE	ERSPEEDING 1	N, GOVERNOR			
	(High	RPM)			
Student				Sheet No.	
Date					
Establish Operating RPM Increase collective					
Reterd throttle					
Maintain operating RPM (with collective and throttle)					
Land at first available safe landing area					
Errors					
Total prior time:	1st trial		Criteria r	eached	

UND	DERSPEEDING N ₂ GOVERNOR (Low RPM)
Student	Sheet No
Date	
Collective: Down to maintain rotor RPM	
Throttle: RETAPD	
Governor switch: EMER	
Governor RPM INC-DECR switch: DECREASE	
Throttle: Advance slowly and firmly to obtain operating RPM	
Errors	
Total prior time:	lst trial . Criteria reached .
	ise that i

	ILS				
Student				Sheet No.	
Date					
	TRANSITI	ION			
Tune radios and identify sta.					_
Track (VOR ±5°) (ADF ±5°) Radar vector (±10°)					
Altitude (±100')					
	ARKER OR INIT	TIAL APPROAC	H FIX		
Turn					
Report					
Altitude (±100')					
	HOLDIN	NG .			
Entry					_
Report					
Heading outbound (±10°)					
Track inbound (*2 1/2°)					
Timing					亡
Altitude (±100')					
	PROCEDURE	TURN			
Pre-landing check completed					
Timing					
Procedure turn altitude					

a.	
	ILS TRACKING INBOUND
Rolled out on course	
Track (*2 1/2°)	
Altitude (±100')	OUTER MARKER INBOUND
Intercept altitude (±100')	
Report	
Airspeed 90 kts (±10 kts)	APPROACH
Localizer (*2 1/2°)	
Glideslope (in the doughnut)	
Min. altitude (-0, +100')	MISSED APPROACH
Time / DH	
Power as required	
Report	
Prescribed procedures	
Errors	
Total prior time:	1st trial : Criteria reached :

Student	TAKEOFF TO HOVER Sheet No
Date	Sheet no.
Collective pitch application (smooth and positive)	
Heading (±5°)	
Drifts (±1')	
Altitude (3' ±1')	
Errors:	
Total prior time:	1st trial : Criteria reached :

	NORM,	AL APPROACH			
Drift (±1')					
Assistance Required					
Approach Termination (to touched Heading (±5°)	lown)				
heading (15)					
Touchdown					
Attitude					
Assistance Required			[]		
Errors:					
Total prior time:	1st trial		Criteri	ia reached	

Prior to takeoff Determine hover power Note OAT Determine power available (Go No-Go) Eakeoff Light on skids properly Power application Smooth				
Determine hover power Note OAT Determine power available (Go No-Go) [akeoff Light on skids properly Power application				
Determine hover power Note OAT Determine power available (Go No-Go) [akeoff Light on skids properly Power application				
Determine power available (Go No-Go) [akeoff Light on skids properly Power application				
(Go No-Go) Takeoff Light on skids properly Power application				
Light on skids properly Power application				
Smooth				
Amount (2% UH-1H)				
Heading (±5°)				
Attitude (40 kts ±10 kts)				
Ground track proper				
Assistance required				
Errors:		1		
Errors: Total prior time: 1st trial	:	Criteria	rea hed	

CTANDARD AUT	T. T. A.	6:-11			
STANDARD AUTO)ROTATION (Stagefield	or Sod Touc	Sheet No.	
Date					
Entry					
Collective pitch Full down					
Trim (±1/2 ball)					
Rotor RPM (in the green)					
Descent Heading (as required)					
Landing area (adequate)					
Airspeed					
Rotor RPM (in the green)					
Landing Approach Deceleration					
Collective pitch (as required)					
Heading (±5°)					
Ground track					
Touchdown Attitude					
Collective pitch (as required)					
Assistance required					
Errors:					
Total prior time:	1st trial		Criter	ria reached	

Date Prior to entry Altitude (±50') Airspeed	STEEP APP	PROACH		Sheet No.	
Prior to entry Altitude (±50') Airspeed					
Altitude (±50') Airspeed					
Airspeed					
	_				
Ground track					
Approach entry Entry point (12-15°)					
Collective pitch reduction					
Approach Descent (12-15°)		7			
Rate of closure					
Approach termination Heading (+5°)					
Touchdown					
Attitude					
Assistance required					
Errors:					
the state of the s	trial [Criteria		
			0.10		-

	SIMULATED A	NTI-TORQUE	FAILURE		
tudent		_		Sheet No.	
ate					
nitially Throttle (manual control)					
Engine RPM (6400)					
Ground track					
Altitude (±100')					
Entry point (5-8°)					
Power use					
Airspe d 60 kts ±5 kts)					
Engine RPM (6400 i.F 6200)					
Descent 5-8°)					
Deceleration					
Engine RrM (6400 L.P. 6200)					
Ground track					
Airspeed (at or above trans- lational lift)					
Throttle usc					
Collective pitch use					

	SIMULATED ANTI-TORQUE FAILURE
Assistance required	
Errors:	
Total prior time:	1st trial : Criteria reached :

APPENDIX E AMPLIFIED CHECKLIST: UH-1D/H

		Dates
BEFOR	E STARTING ENGINE	
1.	Entrance Doors - Secured.	
	Cargo doors may be secured in full open position only if appropriate modifications have been made to the doors and airframe. If a cargo door comes open while in flight, reduce forward spped below 60 knots and secure door.	
2.	Seat and Pedals - Adjust.	
3.	Seat Belt and Shoulder Harness - Fasten and adjust.	
4.	Shoulder Harness Lock - Check.	
5.	Cyclic, Collective, and Throttle Friction - OFF.	
6.	Cyclic, Collective Pitch, and Pedals - Check travel, center cyclic, and pedals. Place collective pitch full down.	
7.	Landing Light and Searchlight - OFF	
8.	AC Circuit Breakers - IN.	
9.	All Radio Equipment - OFF; Set on desired frequencies.	
10.	GOVernor - GOV AUTO.	
11.	DE-ICE - OFF.	
12.	INT AUX FUEL BOOSE Pump - OFF	
13.	LOW RPM AUDIO - OFF or check, spring loaded.	

		, -, -, -, -, -, -, -, -, -, -, -, -, -,
	14.	MAIN FUEL - OFF
(0)*	15.	START FUEL - OFF
	16.	HYDraulic CONTROL Switch - ON.
	17.	FORCE TRIM - ON.
	18.	CHIP DETector Switch - Spring loaded to BOTH.
	19.	Compass Slaving - IN (MAG HDG if applicable).
	20.	Instruments - Check static indications, slippage marks, and operating range limit markings.
	21.	Turn and Slip Indicator - Check race full of fluid.
	22.	Marker Beacon - OFF.
	23.	Clock - Wound and running.
	24.	Magnetic Compass - Check full of fluid and deviation card.
	25.	VSIs - Note indication.
	26.	Heading Indicators - Check RMI selector switch in ADF position, calibration card posted.
	27.	Altimeters - Set to field elevation.
	28.	Airspeed Indicators - Note indication.
	29.	Free-Air Temp Gage - Check reading and condition.
	30.	STARTER GENerator Switch - START.
	31.	NON-ESSential BUS - NORMAL ON.
	32.	VM Selector Switch - BAT (check 24 volts on DC voltmeter), then to MAIN GEN if APU start.
	33.	MAIN GENerator Switch - ON and cover down.
	34.	AC PHASE Selector - AC phase.
	35.	INVTR Switch - OFF.
	36.	Instrument Lights - OFF (Set as desired for night flights).

37.	DC Circuit Breakers - IN, except for armament and special equipment.			1	
38.	PITOT HTR - OFF.		1		
39.	DOME LT - OFF (except for night flight).				
40.	EXternal LTS - OFF (FLASH for night flights as desired).				
41.	ANTI COLLision Light - OFF.		1		
42.	WIPERS - OFF.			1	
	Wipers must not be operated on dry windshield.	 -	+	-	
43.	CARGO RELease Switch - OFF.		-	4	
44.	Cabin Heating Switches - OFF.	-	-	_	
TARTI	NG ENGINE				
1.	BATtery Switch - OFF. (ON for battery start).				
2.	Copilot's Attitude Indicator - Cage (for APU start only).				
3.	INVTR Switch - SPARE (OFF for battery start).				
4.	FIRE DETECTOR Light - TEST (15 seconds maximum).				
5.	RPM Warning Light - ON.				
6.	Fuel Filter and Cargo Release Lights - Press to test.				
7.	Fuel Gage Test Switch - If APU start, depress until fuel quantity drops approximately 200 pounds, then release and check that gage returns to original indication.				
8.	Caution Panel Warning Lights - TEST and RESET master caution light.				
9.	MAIN FUEL Switch - ON (Check fuel pressure [APU START]).				
10.	START FUEL - ON.				
11.	GOVernor RPM INC-DEC Switch - DEC for 10 seconds.				

(0)

Throttle - Check Full Travel and return to flight idle; check operation of engine idle stop, then move throttle to full closed; position throttle as near as possible (on decrease side) of the flight idle stop.
 DOME LT - OFF.
 Fireguard - Posted.
 Rotor Blades - Clear and untied - verbally announce "CLEAR."

16. Start Switch (Trigger) - and Hold; start time; use installed timing device.

Note
During a battery start a minimum of 24 volts should be indicated on the DC voltmeter before attempting start. However, a battery start can be made when voltages less than 24 volts are indicated, provided the indicated battery voltage does not drop below 14 volts with the starter energized.

Caution

Limit starter energize time to 40 seconds. If engine does not start, a 3-minute cooling period is required before beginning another starting cycle. Only three 40-second starting attempts are permissible in any one-hour period.

- (0) 17. Start Fuel OFF at 400°C.
 - 18. Release starter switch at 40% gas producer rpm or after 40 seconds, whichever occurs first.

Caution

During starting or acceleration, the maximum allowable EGT is 760°C. If EGT exceeds 760°C for any period of time, or 650°C (L-13-675°C) for more than 5 seconds, an entry in the 2408-13 is required. If during starting or acceleration it becomes apparent that EGT will exceed 650°C (L-13-675°C) ABORT THE START as follows: throttle full off, fuel system off and continue to motor the starter until EGT decreases.

- 19. Copilot's Attitude Indicator (Battery Start) Cage.
- 20. INVTR Switch (Battery Start) To SPARE.



	21.	Throttle - Slowly advance past engine idle stop to flight idle position. Manually check flight idle stop by attempting to roll throttle off.	
	22.	Gas Producer - 55% to 59% (68% to 72%-L-13). Note	
		A slight rise in N_1 may be anticipated after releasing pressure on twist grip.	
	23.	Engine Oil Pressure - 25 psi minimum.	
	24.	Transmission 0il Pressure - Check normal. Caution If no oil pressure is evident at this time, shut engine down immediately and investigate the cause.	
(N)	25.	All Interior Lights - as desired.	
	26.	Auxiliary Power Unit (APU Start) - Disconnect. Caution Check external power disconnected prior to turning battery switch on.	
	27.	Battery Switch (APU Start) - ON.	
	28.	Fuel Gage Test Switch (Battery Start) - Depress until fuel quantity gage drops approximately 200 pounds, then release and check that gage returns to original indication.	
	ENGI	NE RUN-UP	
		Caution Full forward movement of cyclic may result in damage to short shaft.	
	1.	ICS and Radios - ON as desired.	
	2.	FORCE TRIM Switch - Check in ON position; press release button on cyclic stick to insure proper function; place switch in OFF position and check cyclic and anti-torque controls for freedom of movement and tippath plane for correlation with cyclic movement.	
	3.	HYDraulic CONTrol Switch - Place in OFF position. Check controls for freedom of movement, insure that the collective pitch control is FULL DOWN; then place the switch in the ON position and position the FORCE TRIM Switch ON.	

- 4. Helmet ON.
- FUEL SYSTEM AND DE-ICE FUEL TANK SUMP PUMP circuit breaker out; set CABIN HEAT bleed air selector to position no. 2 (ON if applicable). DE-ICE - ON, Note EGT increase, FUEL TANK SUMP PUMP circuit breaker in, air selector off, DE-ICE OFF. Note fuel pressure returns to normal and EGT decreases. The following is applicable to helicopters serial nos.69-15292 and subsequent: Fuel Boost Pumps - Pull LEFT FUEL BOOST circuit breaker, note caution light on and normal fuel pressure. Pull RIGHT FUEL BOOST circuit breaker, note caution lights on and 0 fuel pressure. After 10 seconds press LEFT FUEL BOOST circuit breaker in, note caution light out, normal fuel pressure. Press RIGHT FUEL BOOST circuit breaker in, all caution lights out. De-Ice/Hot Air-ON, note EGT increase, OFF, note EGT decrease.
- 6. Throttle Slowly increase to full open. Engine rpm (N₂) should stabilize at 6000 plus or minus 50 rpm. Throttle friction as desired.
- PITOT HTR Switch ON. Note loadmeter increase then OFF.
- 8. AC PHASE Selector Check all phases for reading of 115 plus or minus 4 volts; leave in BC.
- INVTR Switch To OFF position check for caution light indication. Switch to MAIN ON check caution light OFF.
- 10. AC PHASE Selector Check all phases for reading of 115 plus or minus 4 volts; leave in the AC position.
- 11. VM Selector Switch Check all positions for indication of 28 to 28.5 volts (27 to 27.5 on standby position); leave in NON-ESSential BUS position.
- 12. Main Generator To OFF position; check caution light indication.
- 13. STARTER GENerator To STBY-GEN position. Main generator loadmeter should indicate "zero" and standby generator loadmeter should indicate a load.

	14.	NON-ESS BUS Switch - Check voltmeter indication of "zero" with nonessential bus switch in NORMAL ON position; NON-ESS BUS switch to MANUAL ON, recheck 27 to 27.5 voltmeter reading; switch to NORMAL ON position.
	15.	VM Selector Switch - Check remaining positions for indication of 27 to 27.5 volts (28 to 28.5 on Main Gen); leave in MAIN GEN position.
	16.	MAIN GENerator - ON and guard closed.
	17.	All Engine and Transmission Instruments - Check for proper indication.
	18.	LOW RPM Switch - AUDIO.
	19.	Governor RPM INC-DEC Switch - Actuate through full range 6000 to 6700 plus or minus 50 rpm. Set rpm at 6600. During governor INC-DEC check, observe low rpm audio and warning light OFF at 6100 +100 engine rpm and/or 300 +5 rotor rpm.
(I)	20.	Communication and Navigation Radios - Perform operational check of all radios and position to ON, as desired; set course selectors as desired.
	21.	Clock - Set.
(I)	22.	Heading Indicator - Indicates plus and 0. Note Refer to Chapter 5 of TM 55-1520-210-10 for pre-flight checks and free gyro mode operation.
(I)	23.	MAG Compass - Corresponds with heading indicator.
(I)	24.	Altimeter - Determine K-factor.
(I)	25.	Attitude Indicator - Set as desired.
	26.	ANTICOLLision Light - As desired.
	27.	FORCED TRIM Switch - As desired for flight.
	28.	Collective Pitch Friction - Check; set as desired.
	PRIOR	TO INSTRUMENT TAKEOFF
(I)	1.	VSI, Altimeter - Indicates climb, descent.
(I)	2.	Turn Needles, Heading Indicator, and Magnetic Compass indicate a Turn to Right - Left.

(1)	3.	Slip Indicator - Ball free in race.	
(1)	4.	Attitude Indicator - Indicates nose high, nose low, bank left, right.	
(I)	5.	Airspeed Indicator - Note indication.	
(1)	6.	ENGine and TRANSmission Instruments - in green.	
(1)	7.	ENGine RPM - As desired.	
(I)	8.	Torque - Note psi for hover.	
(1)	9.	<pre>Index over takeoff heading - Set heading.</pre>	
(I)	10.	PITOT HEAT - ON if necessary	
	BEFORE	TAKEOFF	
		Immediately prior to takeoff, the following checks will be accomplished and announced orally:	<u> </u>
	1.	RPM.	
	2.	Fuel Quantity.	
	3.	Instruments.	
	4.	Caution Lights.	
	5.	Low RPM Audio Warning Switch - AUDIO. Warning Suspend operations immediately if engine or transmission oil pressure and temperature are not within operating limits.	
	6.	Bleed Air Switch - OFF Caution The bleed air heater should be in the OFF position during takeoff and landing and other flight conditions requiring maximum engine power available.	
	ENGINE	SHUTDOWN	
	1.	Collective Pitch - FULL DOWN	
	2.	Throttle - Reduce to flight idle. Check N_1 speed 56% to 58% (70% to 72%, L-13).	
	3.	LOW RPM Audio - OFF after checking operation.	

- 4. FORCE TRIM ON.
- 5. STARTER-GENerator Switch START position.
- (N) 6. EXTernal Lights FLASHING.
 - Exhaust Gas Temp Allow to stabilize (minimum of 2 minutes).

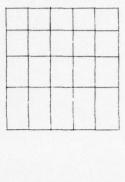
Caution

If a rapid rise in EGT is noted, motor the engine (throttle closed) to allow temperature to stablilze within limits.

- 8. Throttle FULL OFF.
- 9. Main Fuel Switch OFF.
- 10. Radios and ICS OFF.
- 11. All Electrical Switches OFF except main generator and battery.
- 12. Navigation Lights OFF, after rotor is tied down.
- 13. ANTICOLLision Light OFF.
- 14. Battery OFF, after engine tachometer reads
 "zero." (Night, after NAV lights are turned
 OFF.)
- 15. Main Rotor Blades Tie down.
- Conduct a thorough walk-around inspection of the aircraft. (Check oil levels and check for visible leaks.)
- 17. Complete DA Form 2408-12 and -13.

*NOTE:

- (I) Instrument Flight Only
- (N) Night Flight Only
- (0) If Equipment Installed





APPENDIX F

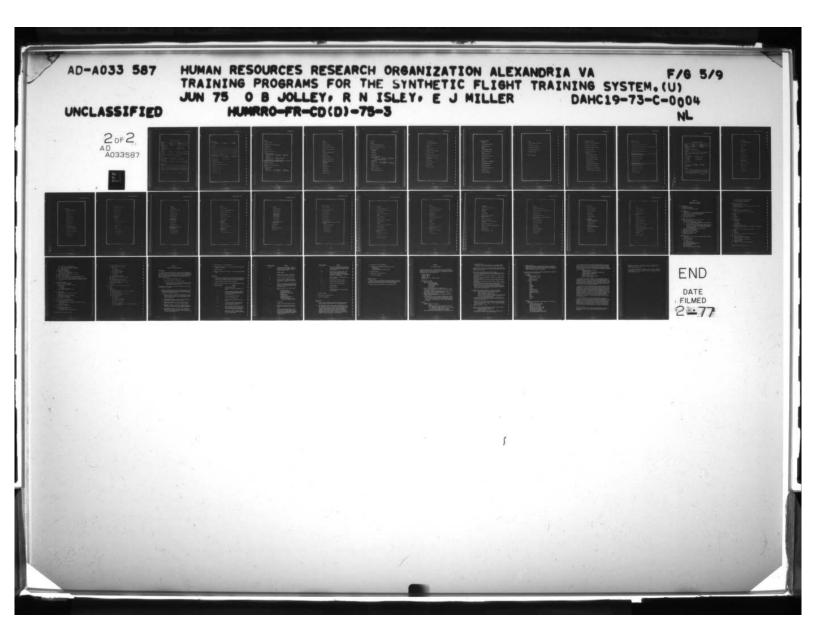
Pre-Solo Checkride Recording Form

Instrument Checkride Recording Form

Advanced Contact Checkride Recording Form

	PRE-SOLO CHECKRIDE R	ECORDING FORM
STUDENT:	RANK:	SSAN:
DATE:	AIRCRAFT:	FLIGHT TIME THIS PERIOD:
INSTRUCTOR:	EXAMINER:	EVALUATION:
WEATHER CONDITIONS:		
Ceiling	Visibility_	
TURBULENCE:		
Calm	Light	Moderate Severe
1. TENSION: Interefe		
2. PREFLIGHT:		hand hand
Preflight Ins	pection	Before Engine Start
Engine Start		Engine Runup
Use of Checkl:	ist	
3. ORAL:		
Aerodynamics	Emerge	ency Procedures
4. BEFORE TAKEOFF CHEC	CK:	
COMMENTS:		
	has/has not demo	onstrated satisfactory working
knowledge of the requir	ed maneuvers/procedur	res and is/is not cleared for solo
flight in the UH-1 heli	copter.	

	•		
	The student	will demonstrate a world-ra begulades	of the fallendar
		will demonstrate a working knowledge procedures on a pre-solo checkride pr	
1	for the first su		
	1.	Preflight Inspection	
	2.	Cockpit Procedures	
	3.	Traffic Patterns	
	4.	Hovering Flight	
	5.	Normal Takeoff/Approach	
	6.	Straight and Level Flight	
	7.	Normal Climb and Descent	
	8.	Level Turns	
	9.	Climbing and Descending Turns	
	10.	Simulated Engine Failure Procedures	
	11.	Deceleration	
	12.	Use of Force Trim	
	. 13.	Hovering Autorotation	
	14.	Standard Autorotation	
	15.	Simulated Anti-Torque Failure	
	16.	Simulated Hydraulic Failure	
	17.	N ₂ Covernor Failure Operations	
21111			



STUDENT:	RANK:	SSAN:	
DATE:	AIRCRAFT:	FLIGHT TIME THIS PERIOD:	_
INSTRUCTOR:	EXAMINER:	EVALUATION:	
WEATHER CONDITIONS:			
Origin		Destination	
Ceiling Vis		Ceiling Vis	
Wind: Velocity	Dir	VelocityDir	
TURBULENCE:			
Calm	Light	Calm	Li
Moderate	Severe	Moderate	Se
ORAL:	<u>'</u>		
Emergency	ATC		
Procedures	Procedure	s Regulations	
FLIGHT PLANNING:			
Publications		F F	igh
rubiteactions	Weather	Form 175 Lo	
Alternate			
COMMENTS:			-
			_
			_
has	has not satisfactoril	y demonstrated necessary skills and	
		y demonstrated necessary skills and instrument rating in the UH-1 helic	орі
			орі
			орі

Instrument (2)

4. EQUIPMENT CHECK: Instruments Communication Navigation Equipment Equipment
5. CLEARANCE:
6. DEPARTURE PROCEDURES: (To primary airway) Instrument Set-up Transponder Frocedures Before Takeoff Check
Communications SID Procedures
Aircraft Control:
Climb (500 fpm ±100)
Track (VOR ±2°, ADF ±5°, Heading ±10°)
Altitude (±100')
Airspeed ±10 kts
Trim (thy ball)
Level Off Check

7.	ENROUTE:
	Altitude ±100'
	Track (VOR ±5°, ADF ±5°, Heading ±10°)
1	Reports
	ETAs (t3 mins)
	Airspeed (90 kts ±10 kts)
	Trim: Ball centered 2's
	Holding: Entry Timing Pattern
	Procedures:
	Position Fixing Lost Communication
1_	Communication Use of Co-pilot
8.	APPROACH: VOR ±2° mark left side; ADF ±5° mark right side
	Pre-landing Check
	Vector to Final Course (Yes - No)
	 .
	Procedure Turn:
	Published ±100'
	Distance Altitude Pattern

Instrument (4)

B

Inbound:
Track
Rate of Deacent
Airspeed (90 kts ±10 kts)
Low Station Altitude (±100')
Final:
Report (Low station inbound)
Track
Rate of Descent
Airspeed (90 kts ±10 kts)
Min. Altitude (-0, +100')
Missed Approach:
Ti=
Power Proper
Report
Procedure
Trim (24 ball)

Instrument (5)

9. ILS APPROACH:
Transition:
Tune Radios and Identify Station Track (VOR ±5°, ADF ±5°, Vector ±10°)
Altitude (±100')
Outer Marker:
Turn
Altitude (2100')
Holding: Entry Timing Pattern
Pre-landing Check Completed Procedure Turn:
Distance Published ±100' Pattern Inbound:
Track (±2½°)
Altitude (±100°)
Airspeed (90 kts ±10 kts)

	Instrumen
Outer Marker Inbound:	
Intercept Altitude (±100')	
Report	
Approach:	
Localizer (±2½°)	
Glideslope (In the doughnut)	
Trim (2½ ball)	
Minimum Altitude (-0, +100')	
Airspeed (90 kts ±10 kts)	
Missed Approach:	
Time/DH	
Power Proper	
Procedure	
Trim (±15 ball)	

(6)

1

10. GROUND CONTROLLED APPROACH
Initial Call:
Communication Procedures
Position Report
Transition:
Transponder Procedures
Heading Control (±10°)
Altitude (±100')
Airspeed (90 kts ±10 kts)
Trim (th ball)
Pattern:
Rolled out on Heading (±10°)
Heading Control (210°)
Altitude (2100')
Airspeed (90 kts ±10 kts)
Arisped (50 kts 110 kts)
Trim (th ball)
Pre-landing Check

	-
]
	1
Final:	
Turn (Rolled out on Heading ±10°)	
Keading Control (25°)	
Glideslope Satisfactory	
Airspeed (90 kts ±10 kts)	
Trim (2½ ball)	
, ——,	
Voice Procedures	

Instrument (8)

11. ENGINE FAILURE (During Flight):
Rotor RPM Maintained (Collective Down)
Ball Centered (14 ball)
Glide Established 80 kts (±10 kts)
Called out RPM/N ₁ /AS
Heading Maintained (±10°)
If time allows:
Transponder set to EMER
Transmit MAYDAY (3 times)
Transmit A/C Identification
Transmit Position
Transmit Nature of Distress
Main Fuel: OFF (Verbal Only)
Battery Switch: OFF (Verbal Only)
Shoulder Harness: LOCK (Verbal Only)

Instrument (10)

12. HYDRAULIC POWER FAILURE:
Adjust Airspeed
Hydraulic Control Circuit Breaker: OUT
Hydraulic Control Circuit Breaker: IN (If Power Not Restored)
Hydraulic Control Switch: OFF
Master Caution Warning Light: RESET
Landing: Accomplish 13. UNUSUAL ATTITUDES (Steep Climbing Turn)
Wings: Level Attitude
Pitch: Level Attitude
Power: Adjust
Trim (±4 ball)
14. UNUSUAL ATTITUDES (Steep Descending Turn)
Wings: Level Attitude
Pitch: Level Attitude
Power: Adjust
Trim (1/2 ball)

STUDENT:	RANK:	SSAN: FLIGHT TIME
DATE:	AIRCRAFT:	THIS PERIOD:
INSTRUCTOR:	EXAMINER:	EVALUATION:
	Visibility	
. TENSION: Inter	rfered with performance	No Yes
. PREFLIGHT:		
	light ection	Before Engine Start
Engir Star	ne	Engine Runup
Use of Check	of klist	
. ORAL:		
Aero		rgency Operators cedures Manual
. BEFORE TAKEOFF	CHECK:	
COMMENTS:		
quired for qua	has/has not demor	nstrated necessary skills and knowledges t licopter.
	Pilot	Checkpilot

5. TAKEOFF TO HOVER: Collective Pitch Application (Smooth and Positive)
Collective Pitch Application
Collective Pitch Application (Smooth and Positive)
Heading (± 5°)
Drift
Hovering Altitude
6. HOVERING FLIGHT:
Altitude
Heading (±5°)
Heading (1)
Hovering Speed
Turns
7. LANDING FROM A HOVER:
Use of Collective Pitch
Heading (±5°)
Descent
Drift
Touchdown

8. NO	ORMAL TAKEOFF:
a.	. Through Effective Translational Lift
	Pre-Takeoff Check
	Clearing Turn
	Light on Skids
	Power Application
	Acceleration
	Heading (±5°)
	Trim (±½ ball)
ь	. After Effective Translational Lift
	Airspeed (60 kts ±10 kts)
a -	Rate of Climb (500' ±100')
	Ground Track
	Trim (th ball)
1	

Advanced Contact (4)

9. TRAFFIC PATTERN:
Airspeed (90 kts ±10 kts)
Altitude(±100')
Ground Track
Ground Track
Spacing
Trim (±½ ball)
10. NORMAL APPROACH:
a. Base Leg
Turn
Deceleration (60 kts ±5 kts)
Descent
Ground Track
Turn
Altitude (±50°)
Airspeed (60 kts ±5 kts)
Ground Track

Contraction of the second seco
c. Approach Entry
Entry Angle
Lancity August
Collective Pitch Reduction
d. Approach
b
Approach Angle
Rate of Closure
Thate of Closure
Ground Track
Trim (±4 ball)
e. Approach Termination
in in the second
Heading (±5°)
neading (23)
Touchdown
Attitude
Use of Collective
11. MAXIMUM PERFORMANCE TAKEOFF (40 kts or 100')
a. Prior to Takeoff
a. Frior to lakeoff
Pre-Takeoff Check
Determine Hover Power
GO-NO GO Check

Advanced Contact (6)

b. <u>Takeoff</u>
Light on Skids
Power Application (Smooth and Proper)
Power Increase (2% UH-1H)
Heading (±5°)
Attitude (40 kts ±10 kts)
Ground Track
Trim (24 ball)
12. STEEP APPROACH:
a. Prior to Entry
Altitude (±50')
Airspeed (60 kts ±5 kts)
Ground Track b. Approach Entry
Entry Angle
Collective Pitch Reduction

c. Approach
Approach Angle
Rate of Closure
Ground Track
Trim (th ball)
d. Approach Termination
Heading (±5°)
Touchdown
Touchdown
Attitude
Use of Collective Pitch
13. HOVERING AUTOROTATIONS:
Heading (±5°)
Drift
Use of Collective Pitch
Touchdown
A CONTRACTOR OF THE PARTY OF TH

Advanced Contact (8)

14. HYDRAULICS OFF LANDING:
Altitude (±100')
Published Emergency Procedure
Pre-Landing Check
Approach Angle
Rate of Closure
Ground Track
Touchdown
Collective Braking
Trim (24 ball) 15. SIMULATED ANTI-TORQUE FAILURE (L = Left Pedal, R = Right Pedal)
a. <u>Initially</u>
L R Throttle (Manual Control)
RPM Control (6400)
Ground Track
Altitude (±100')
Pre-Landing Check

	b. Approach Entry
	LR
	Airspeed (60 kts ±5 kts)
	Engine RPM (L.P. 6200, R.P. 6400)
	Entry Angle
	Entry Angle
	Collective Pitch Reduction
	c. Approach
	Approach Angle
	Approach Angle
	Rate of Closure
	Nace of Vaccount
	Engine RPM (L.P. 6200, R.P. 6400)
	Ground Track
	d. Touchdown
	1033333
	Airspeed (at or above translational lift)
	Lane Alignment
	(
	Touchdown
	I—————————————————————————————————————
	Collective Braking
16.	STANDARD AUTOROTATION:
	a. Entry
	Collective Pitch Full Down
	Trin
	Rotor RPM (in the green)
	Throttle (flight idle)
	inroctie (ringet idie)

b. <u>Descent</u>
Ground Track
Glide Established (80 kts ±10 kts)
Rotor RPM (in the green)
Trim
c. Landing Approach
Deceleration
Collective Pitch (as required)
Ground Track
Trim
d. Touchdown
Heading
Attitude
Collective Pitch (as required)
17. LOW LEVEL AUTOROTATION: a. Entry
Collective Pitch Full Down
Trim
Deceleration

b. Landing Approach
Collective Pitch (as required)
Collective Fitch (as required)
Ground Track
c. Touchdown
Heading
Attitude
Collective Pitch (as required)
18. ACCELERATION & DECELERATION:
a. Entry
Altitude (min 500' AGL)
Airspeed (90 kts ±10 kts, roof mtd. pitot)
b. Deceleration
Power Reduction (10 lbs ±1 lb)
Total Reduction (10 to 17 to 17
Pitch Attitude: Adjust to Maintain Altitude (±50')
Trim (24 ball)
Heading (±10°)
Airspeed (40 kts, -0 + 5kts)
The state of the s

The state of the s
c. Acceleration
Neces action
Power Increase (12 lbs ±1 lb)
Pitch Attitude: Adjust to Maintain Altitude (±50')
Trim (±½ ball)
Heading (±10°)
Airspeed (90 kts, ±10 kts)
Power Reduction: As Required to Maintain Airspeed 19. CONFINED AREA OPERATIONS:
High Reconnaissance
Pre-Landing Check
Low Reconnaissance
Approach
Touchdown
Ground Reconnaissance
Pre-Takeoff Check
GO-NO GO Check
Takeoff

20.	PINNACLE OPERATIONS:
	High Reconnaissance
	Pre-Landing Check
	Low Reconnaissance
	LOW RECORDARY STATES
	Approach
	Touchdown
	Pre-Takeoff Check
	Takeoff
21.	SLOPE OPERATIONS:
	a. Descent to Touchdown (Upslope Skid)
	Descent (Vertical)
	Heading (±5°)
	Use of Collective
	b. Descent to Touchdown (Downslope Skid)
	Heading
	Use of Collective
	Use of Cyclic

Advanced Contact (14)

c. Takeoff
Use of Cyclic
Power Application
Heading (±5°)
Ascent (Vertical)
22. FORCED LANDING:
a. Entry
Collective Pitch Full Down
Trim (±4 ball)
11111 (-7 0411)
Rotor RPM (in the green)
motor with (In the Steen)
Glide Established
Glide Established
Checked RPM/N ₁ /AS (called out)
b. Descent
Selection of Landing Area
Emergency Communication Procedures (if time permits)
Airspeed (as required)
Made the Field
c. Termination
(1) Power Recovery
(2) Climb

APPENDIX G

MISSION MODEL: PLAN, FILE AND FLY AN IFR MISSION

- 1. Check weather:
 - a. Can flight be made?
 - b. Will it be IFR or VFR?
 - c. If IFR, is an alternate required?
- 2. Plan flight:
 - a. Check NOTAMS
 - b. Determine that all required publications, charts, approach plates, computers, etc., are available to the pilot
 - c. Determine data and fill in the flight log
 - d. Fill in flight plan (175)
 - e. Get weather briefing (175-1)
 - f. File 175 and 175-1
- 3. Preflight, start, and run-up
 - a. Check aircraft status. Forms 2408-12 and -13
 - (1) Note discrepancies
 - (2) Refueling accomplished and recorded
 - (3) Due date of next periodic (is sufficient time available to allow completion of flight?)
 - (4) Is the aircraft flyable? Do the forms indicate that all systems (radio, navigation equipment, etc.) required for successful completion of this flight are available and functioning normally?
 - (5) Make the appropriate entries in Form 2408-12
 - b. Exterior check (use checklist)
 - c. Interior check (use checklist)
 - d. Before start check (use checklist)
 - e. Start (use checklist)
 - f. Engine run-up (use checklist)
 - g. Engine health check
- 4. Prepare to hover
 - a. Tune and copy Automatic Terminal Information Service (ATIS) if available
 - b. Contact Ground Control
 - (1) Taxi clearance
 - (2) Time
 - (3) Active runway
 - (4) Altimeter setting
 - (5) Clearance delivery instructions
 - c. Contact Clearance Delivery
 - (1) Copy clearance
 - (2) Read back
 - (3) Set navigation radios as desired

- (4) Set communication radios as desired
- (5) Set course selector as desired
- (6) Set transponder as desired
- 5. Hover to takeoff area:
 - a. Raise the helicopter to a hover
 - b. Initiate air taxi
 - c. Perform "prior to instrument takeoff" check (use checklist)
 - d. Land at takeoff area
- 6. At takeoff area:
 - a. Contact the tower for takeoff clearance
 - b. Move to the takeoff position
 - c. Perform "before takeoff" check (use checklist)
- 7. Instrument takeoff
 - a. Power
 - b. Trim
 - c. Attitude
 - d. Airspeed
- 8. At climb airspeed
 - a. Establish climb
 - (1) Power
 - (2) Trim
 - (3) Attitude
 - (4) Airspeed
 - b. Report airborne
 - c. Make other required reports
 - Follow other instructions, if any (transponder, radio contact, reports, etc.)
- 9. At assigned altitude (establish cruise altitude)
 - a. Power
 - b. Trim
 - c. Attitude
 - d. Airspeed
 - e. Check engine instruments for normal indications
- 10. First leg of flight
 - a. Intercept and maintain course
 - b. Prepare to compute fuel consumption
 - (1) Record time
 - (2) Record amount of remaining fuel
 - *c. Maintain altitude
 - *d. Maintain airspeed
 - e. Prepare to verify ETA (as soon as possible)

^{*}These duties are performed on all legs.

- (1) Record time at known location
- (2) Record time at second known location
- (3) Determine distance between two locations
- (4) Compute ground speed
- *f. Revise ETA if necessary
- g. Compute fuel consumption
 - (1) Hasty check at 10 minutes
 - (2) Normal check after minimum of 15 minutes
- *h. Monitor engine instruments
- *i. Remain alert for unusual sounds or vibrations
- *j. Maintain communications
- *k. Monitor weather broadcast for changes in enroute weather
- *1. Monitor enroute weather for unforecast conditions
- *m. Report any unforecast weather conditions encountered to controlling agency (Center, Flight Service Station, Approach Control, etc.)
- 11. Approaching checkpoint
 - a. Prepare to identify checkpoint
 - b. Prepare report, if required
 - c. Prepare for required changes in flight
 - (1) Heading
 - (2) Altitude
 - (3) Holding
 - (4) Navigation procedures
 - d. Make frequency changes, if required
- 12. Over checkpoint
 - a. Record time
 - b. Turn to intercept new course, if required
 - c. Verify ETA
 - d. Update ground speed, if necessary
 - e. Make any required changes in report
 - f. Tune radios, if necessary
 - g. Make report, if required
- 13. Second, and succeeding flight legs
 - a. Intercept and maintain new course
 - b. Update flight log
 - Continue with those duties marked with an asterisk on first flight leg
 - d. Verify ground speed
 - e. Monitor fuel consumption
- 14. Approaching destination
 - a. Turn controls over to copilot
 - b. Study approach plate
 - (1) Minimums for available approach(es)

^{*}These duties are performed on all legs.

- (2) Transitions to final approach fix
- (3) Field elevation
- (4) Procedure turn (if applicable)
 - (a) Type
 - (b) Altitude
 - (c) Outbound heading
 - (d) Inbound heading
- (5) Radio frequency: UHF-VHF
 - (a) Approach Control
 - (b) Tower
 - (c) Ground
- (6) Missed approach procedures
- (7) Minimum sector altitudes
- (8) Time from final fix to field boundary
- c. Brief copilot
 - (1) Make appropriate radio calls
 - (2) Information relative to approach
 - (3) Pre-landing check
 - (4) Notify pilot when ground is sighted
- d. Take control from copilot
- e. Tune radios
- 15. Over destination (approach: VOR, ADF, GCA, ILS)
 - a. Time
 - b. Turn
 - c. Report
 - d. Begin approach
 - (1) Time outbound leg
 - (2) Descend to procedure turn altitude, if applicable
 - (3) Make procedure turn
 - (4) Turn inbound on final
 - (5) Descend to final approach fix altitude
 - (6) Before landing check
 - (7) Final approach fix
 - (a) Check time
 - (b) Report
 - (c) Descend to minimums
 - (d) Time to field boundary
 - (e) Report VFR to tower
 - (f) Land
 - (g) Taxi to tie-down area
 - (h) Close flight plan
 - (i) Make appropriate entries in Forms 2408-12 and -13
 - (j) Secure aircraft

APPENDIX H

OUTINE OF 2C35 TRAINING PERIODS

First Period

1. The student copilot (SCP) will read each checklist CHALLENGE part of the item (all before the dash), hesitate, then read the RESPONSE. The instructor pilot (IP) should locate the item for the student pilot (SP). The SP should touch the item, indicating to the instructor that he has located it, repeat the CHALLENGE, perform the function, then repeat the RESPONSE, e.g.:

SCP: "Hydraulic Control Switch - ON"

IP: Locates hydraulic control switch for SP

SP: Places his hand on the hydraulic control switch, checks that the switch is in the ON position, then responds "Hydraulic Control Switch - ON."

This procedure is followed until the student demonstrates that he knows the location of all controls.

- Students will perform all items on the following checks: (NOTE: the pattern and sequencing of the check prior to the student's first performance should be discussed. The IP will also explain the purpose of each check as noted.)
 - a. Prestart. Purpose: to insure that the pilot and copilot are properly seated and secured, and that applicable aircraft system controls are in the proper operating mode prior to starting the engines.
 - b. Starting engine. Purpose: provides a safe, standardized procedure for starting the aircraft engine.
 - c. Engine run-up. Purpose: to determine that the aircraft engine and applicable systems are operating normally.
 - d. Pre-takeoff. Purpose: a final check, made just prior to takeoff, to insure that the aircraft engine and applicable systems are operating normally as required for flight.
 - e. Engine shutdown. Purpose: provides a safe, standardized procedure for stopping the engine, placing applicable systems in the appropriate inoperative status, and securing the aircraft after operation.

- 3. Repeat Items 2a. 2e. with second SP.
- 4. Explain Forms 2408-12 and -13 to the students and have them make appropriate entries.
- 5. Debrief students.
- Make study assignments for Period 2: -10 checklist and programed texts.

Second Period

- 1. Brief students: designate who will act as SP for first trial and what will be covered during the period; check Form 2408-13 for aircraft status and make appropriate entries on Form 2408-12.
- 2. Proceed through the checklist items as in Period 1.
- 3. Introduce systems information for the following checklist items:

Checklist Number	Comment
	BEFORE STARTING ENGINE
6	Function of the cyclic, collective, and pedals
19	Difference between IN and OUT. When to place in OUT position; what capabilities are available in terms of Free Gyro or turn indicator use
20	Review instrument panel clusters
25	How to set with screwdriver
26	Purpose of two positions and #1 and #2 pointers
27	Set to field elevation; when engine is started may change. Leave alone; will indicate properly in the air
30	Start - generator serves two functions. Standby generator - engine driven, starter battery or APU powered. Start position S/G acts as starter to start aircraft.
32	Important that battery provides maximum voltage for starting. Precludes hot start due to low turbine RPM.

Checklist Number Comment 35-44 To prevent drain on battery. Explain primary and secondary lights. Cabin heat switches--explain if engine temperature surges turn bleed air OFF STARTING ENGINE 1 Provide power to crank the starter if battery start; otherwise APU does this 11 Positions solenoid to minimum setting 16 Battery start check minimum 14 volts. T-13 engine goes to 40% in approximately 20-25 seconds. Demonstrate hot and hung start. 19 Hold in caged position until the inverter switch is in SPARE (prevents damage to the instrument, caused when instrument erects and stops are hit). 20 Which instruments are lost when inverter fails: AC Failure Relay Fuel Quantity Indicator Attitude Indicators AN/ASN 43 (Gyro Mag Compass) Transponder Course Indicator Torque Pressure Transmission Oil Pressure Engine Oil Pressure Fuel Pressure (remains at last position until driven to "0") ENGINE RUN-UP 1 Purpose of the switch and how to check What hydraulic boost does for controls, necessity for doing check the way it is How fuel pressure system works. One or two electrically driven fuel pumps, one in each tank, one engine-driven. Enginedriven will not supply sufficient pressure

pressure altitude.

to keep engine running above 4600 feet

Checklist Nu	<u>Comment</u>
7-9	Check is to determine that each inverter is supplying appropriate AC voltage (115V) to each phase of the three-phase 115V AC electrical system
10-15	This check is to determine that in the event of main generator failure, the standby generator picks up the load as it should. In addition, the non-essential bus switch is functioning
22	How to synchronize the compass with the sensing element
23	Mag compass is the master compass
24	Purpose of the altimeter setting
25	Purpose of pitch and roll knobs
	ENGINE SHUTDOWN
2	Why this is done
6	Hot shutdown (must be in start position to motor engine

- 4. Debrief students.
- 5. Make entries in 2408-12 and -13.
- 6. Give study assignments.

Third Period

- 1. Brief students: Designate SP and SCP and what will be covered during the period; check 2408-12 and -13 and make entries.
- 2. Proceed through checklist items using the -10 checklist. SCP will challenge only, and SP will answer with response after he has performed the task. In order that the IP will know that he is checking the appropriate switch, instruments, knobs, etc., the SP will touch or point it out to the IP as he makes the response. If SP does not respond properly, and it becomes apparent he does not know what to do, SCP will tutor him as to the appropriate action to take. If SCP does not know, IP will explain. IP will record errors on form provided.

- 3. Practice hot start and hung start.
- 4. Demonstrate the following new emergency procedures: Low battery Low oil pressure N₂ governor failure, high and low side
- 5. Debrief students.
- 6. Make entries in 2408-12 and -13.
- 7. Give study assignments.

Subsequent Periods

Practice the above until students can make the appropriate response to starting and run-up malfunctions and accomplish the pre-start, start, run-up, and shutdown procedures with one error-free performance.

Debrief and make study assignment for first period in 2B24.

APPENDIX I

2B24 TRAINING PERIODS

For early training periods, the following outline has been developed. During these periods, the SP will occupy the pilot seat, the IP the copilot seat, and the SCP the instructor's seat. Employment of the freeze capability should be particularly noted.

1. Initial Conditions:

Altitude: 3000' Airspeed: 80 KTS Heading: 360°

Control parameters: Everything frozen

Motion: On

- 2. Heading (roll control)
 - a. Attitude indicator

Explain: Sky representation
Ground representation
Horizon representation
Roll trim knob

Pitch trim knob Bank indices

b. Cyclic control (unfreeze lateral parameter only) Explain relationship between bank indices and lateral movement of the cyclic control.

Allow student to practice establishing various angles of bank (selected by the IP) to acquaint him with

c. Radio magnetic indicator (RMI) heading indicator (unfreeze) Explain purpose and operation.

Explain relationship between heading, bank angle, and lateral cyclic control.

Have student turn to selected headings, right and left, using 7° and 14° angles of bank. Point out differences in rate of turn, and that headings increase in right turns and decrease in left turns. Use heading bug. Student practices maintaining heading.

- 3. Turns
 - a. Standard rate

Explain: Turn needle (unfreeze)

Purpose

Relationship between cyclic control and turn needle How to establish a standard rate turn:

- (1) Establish 140 angle of bank on attitude indicator
- (2) Make fine adjustments with turn needle Relationship between bank angle and airspeed

Have student practice making standard rate turns right and left.

Half standard rate
 Explain rate of turn, when used (20° or less heading change).
 Have student practice half standard rate turns right and left.

Have student practice turns. IP selects heading. Select some headings 20° or less to give sutdent practice in determining when to make half standard rate turns.

Pitch control (unfreeze fore and aft control only)

Explain the relationship between fore and aft movement of the cyclic control and the vertical movement of the aircraft symbol on the attitude indicator.

Have student practice making attitude changes of 1, 2, and one-half bar widths.

Explain the relationship between pitch attitude and airspeed (pitch controls airspeed).

Have student practice maintaining and changing airspeed (disregard altitude).

Explain the relationship between pitch attitude: changes, airspeed, and altitude.

Have student practice making small altitude changes (less than 100', or speed changes of 10 KTS or less) with pitch changes.

Unfreeze lateral control and have student practice making turns to selected headings while maintaining airspeed (point out altitude loss). Next, have him make some turns and maintain altitude with pitch (point out airspeed loss).

5. Power control (unfreeze collective, freeze cyclic control)
Explain: Relationship between power, airspeed, and altitude
Power controlled by collective pitch control
Power is referred to as "torque"
Unit measurement of torque is "pound"
Amount of torque being used is indicated on the Torque Meter
One pound of torque equals two knots or 100 FPM rate of
altitude change (approximately)

Have student practice making power changes and, to familiarize him with the Torque Meterand its location, have him call out the indications on the Torque Meter.

Unfreeze pitch control and have the student practice making airspeed changes while maintaining constant altitude, and altitude changes while maintaining constant airspeed. Stress: pitch for airspeed; power for altitude.

Explain use of power in turns.

6. Trim (ball centered--longitudinal axis and line of flight parallel)
Explain: Relationship between power and ball
Pedals control ball position
Power and heading

Have student practice making power changes while keeping the ball centered with the pedals. Cross-check heading.

7. Force trim button Student should remember system information learned in the 2C35. Explain practical use in the 2B24 and insure that the student understands and practices using the force trim button. Once he understands the principles, its use is optional.

At this point in his training, the student should be able to demonstrate the following knowledges and skills:

Knowledges. Verbalize the functioning of: Cyclic control Collective pitch control **Pedals** Force trim button Define the following: Pitch Rol1 Yaw Trim Heading Altitude Airspeed Torque Attitude Standard rate turn Vertical velocity Standard rate climb Standard rate descent Bank Ball Nose Needle Force trim Pounds Bar

Skills.

Respond, with appropriate control movement, to the following verbal cues:

Raise the nose one bar high
Lower the nose one bar low
Establish 15° right bank
Establish 15° left bank
With pedals, move the ball right
With pedals, move the ball left
Increase power five pounds
Decrease power five pounds
Trim cyclic to right with force trim
Trim in right pedal with force trim

The student will have had no flight training previous to flying the 2B24. In order to achieve the skills needed for mission-concept flight training, he will first need to develop the skills necessary to coordinate all the controls required for minimum aircraft control. Toward this end, some practice flying of the following control coordination exercises will be necessary:

Maintaining heading and altitude
Changing airspeed
Changing heading, standard and half standard rate
Changing altitude, standard rate
Changing heading and altitude
Instrument takeoff
Ground controlled approach ASR-PAR
Unusual attitude recovery

It is not necessary that the student learn to perform these exercises to a high degree of proficiency. However, he should develop sufficient knowledge and skill to enable him to perform the flight control functions necessary to maintain reasonable control of the attitude of the aircraft while learning advanced navigation, radio, and approach procedures. His proficiency in aircraft control will improve as a result of the flying practice obtained learning advanced procedures.*

At this point, each training period should follow the Course Outline in Appendix B, and, where appropriate, Maneuver Performance Record Forms (MPRFs) will be used to control the amount of practice provided for each maneuver and to provide detailed and systematic feedback to the students. The information recorded on the MPRFs will also provide data which, when submitted to statistical analysis, will provide information to help evaluate the program. In addition to those covering the control coordination exercises, MPRFs will also be used to score student performance on advanced navigation, approaches, and selected emergency procedures. Throughout this training, all relevant communications and reporting procedures will be practiced by each student in accordance with all simulator training tasks.

When the student has met appropriate MPRF criteria, he should be introduced to VOR orientation, track interception, and tracking. At this time, the mission training concept can be employed in simple terms. To introduce the student to VOR orientation, the task should be made into a problem such as: "Locate the aircraft in relation to the Dothan VOR. Determine the radial the aircraft is on, track to the station on that radial, and proceed to Marianna VOR on V- ." This is a simple mission, and the student can be worked through it by a process of verbalization,

^{*}Specific performance criteria cannot be specified in detail until the program described here can be tested using an appropriate student population.

discussion, and practice. As an aid, it should be remembered that the device can be frozen at any time while a point is explained and a relaxed discussion takes place.

As the student progresses, the mission can be expanded. Additional tasks, reporting, holding, an occasional emergency, computing ETAs and fuel consumption, etc., can be added. Early in the program, preparation of flight plans must begin.